

Keynote Lecture: Galileo

Marco FALCONE – European Space Agency

IGS Workshop 2017, Paris – July 3-7 2017

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Constellation Satellites





S/C Prime Contractor Astrium GmbH (now Airbus Defence & Space) <u>4 satellites – <mark>4 In-Orbit</mark></u>

Mass at Launch Power Consumption Dimensions Orbit Injection Attitude Profile 700kg 1420W 2.7 x 1.6 x 14.5 m Direct into MEO orbit Yaw Steered



S/C Prime Contractor OHB Systems GmbH P/L Prime Contractor SSTL Ltd 22 satellites – **14 In-Orbit**

Mass at Launch Power Consumption Dimensions Orbit Injection Attitude Profile 733kg 1900 W 2.5 x 1.1 x 14.7 m Direct into MEO orbit Yaw Steered

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Current FOC1 constellation: 18 spacecraft





Constellation Status (30 June 2017)



Constellation Status (30 June 2017)



Constellation Status (30 June 2017)



Launch 8 In-Orbit Testing (IOT)











As-observed Ranging Performance

Ranging Accuracy including L8 Satellites



L8 Satellites

DF 100 min prediction accuracy																		
GSAT	0210	0211	0212	0213	0214	0207	0208	0209	0101	0102	0202	0201	0103	0204	0205	0203	0206	
SVID	E01	E02	E03	E04	E05	E07	E08	E09	E11	E12	E14	E18	E19	E22	E24	E26	E30	Cumulative
Run 1 DF OS (E5a-E1)																		
Orbit RMS error @WUL [m]	0.37	0.46	0.38	0.39	0.54	0.33	0.39	0.42	0.37	0.38	3.01	3.13	0.27	0.33	0.43	0.43	0.35	1.12
Clock RMS error [m]	0.24	0.26	0.25	0.27	0.25	0.19	0.26	0.30	0.22	0.24	0.23	0.27	0.15	0.27	0.19	0.31	0.24	0.25
Ranging 95%-ile error @WUL [m]	0.75	1.04	0.86	0.92	1.05	0.79	0.87	1.00	0.67	0.70	1.34	2.88	0.55	0.96	0.85	0.96	0.78	0.93
Ranging 95%-ile error @AUL [m]	0.56	0.70	0.62	0.61	0.74	0.53	0.66	0.75	0.40	0.50	0.92	1.71	0.37	0.66	0.56	0.75	0.58	0.68
Run 2 DF OS (E5b-E1)																		
Orbit RMS error @WUL [m]	0.33	0.42	0.35	0.36	0.56	0.34	0.35	0.38	0.36	0.36	2.99	3.12	0.27	0.34	0.39	0.39	0.32	1.10
Clock RMS error [m]	0.19	0.22	0.23	0.22	0.21	0.16	0.21	0.26	0.22	0.24	0.21	0.21	0.26	0.28	0.17	0.27	0.22	0.23
Ranging 95%-ile error @WUL [m]	0.63	0.92	0.80	0.82	1.02	0.73	0.76	0.91	0.67	0.71	1.25	2.49	0.67	0.96	0.75	0.86	0.70	0.85
Ranging 95%-ile error @AUL [m]	0.46	0.58	0.54	0.53	0.70	0.49	0.55	0.64	0.43	0.52	0.82	1.58	0.51	0.67	0.46	0.67	0.49	0.61

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Positioning Performance & Availability Horizontal accuracy of global stations on a Hor.Pos.Error = 3.457 m (95%) sample day: 4 more satellites operational: in May/June 2017 * Ξ Satellites in operational constellation 11 \rightarrow 13 $\rightarrow 15$ * ANorth Availability of H Accuracy <10 m $89\% \rightarrow 96\% \rightarrow 98\%$ w/o L8 satellites: Global PDOP <=6 availability $41\% \rightarrow 60\% \rightarrow 78\%$ * 3.46m (95%) -10 -8 -6 -4 -2 0 2 4 6 8 AEast [m] Hor.Pos.Error = 2.857 m (95%) **Global Satellite visibility March 2017 Global Satellite visibility March 2017 Global Satellite visibility April 2017** FNAV FNAV (incl. GSAT0207, GSAT0214) FNAV (incl. L8 SV E07, E03, E04, E05) 100% 90% 904/ 50% >=N 80% 80% ROW Occurrance probability aged over location and 70% 70% 70% Ξ 60% 60% 60% ANorth 50% 50% 50% Occurrance aged over I 40% 40% 40% 30% 30% 30% Occur 20% 20% 20% ver 10% 10% 10% 0% with L8 satellites

2

3 4 5 6 7

Visible number of satellites (NumSV)

2.53m (95%)

AEast [m]

0 2 4 6 8 10

-6 -4 -2

-10 -8

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Visible number of satellites (Num5V)

18

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Visible number of satellites (NumSV)

Positioning Performance

- Comparison:
 - \rightarrow w/o L8 s/c
 - \rightarrow with L8 step1 (GSAT0207, GSAT0214)
 - \rightarrow with L8 step2 (GSAT0212, GSAT0213)
- Constraint: PDOP <=6
- 3-D Positioning FNAV dual frequency
 - H Acc Mean: 2.96 m \rightarrow 2.62 m \rightarrow 2.29 m
- 3-D Positioning FNAV single frequency
 - H Acc Mean: 16.80 m → 15.06 m → 13.39 m



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PVT Field Testing











Galileo-GPS Time Offset





Galileo-GPS Time Offset



GST-GPS Time Offset Message Availability



★ GGTO availability improved over the last year





esa

NeQuick G performance

NeQuick Correction Performance



- Correction performance defined as the remaining ionospheric delay after correction with the Nequick model (Jan - Jun 2016 1Hz data)
- 75 to 80% average ionospheric model correction capability (very good behavior in equatorial zones)



GSAT0201/0202 Orbit



VS09 Orbit injection anomaly left GSAT0201/0202 in highly eccentric orbits



- Both spacecraft safely raised to higher orbit
 - ✓ Perigee raised from 13700 to 17200 km
 - ✓ Eccentricity reduced from 0.23 to 0.15
 - ✓ Above Van Allen belts & Earth Sensors operational range
- Broadcasting of Dummy Messages since 2015 to support scientific experimentation
- Broadcasting navigation message since August 2016 for testing

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L3 Ranging Performance Improvement Test



GSAT0201 FNAV Dual Frequency SISE Global Average (January - March 2017)



/NAV Ranging Performance 15 th percentile	Worst Nominal S/C	Worst L3 S/C		
Before optimisation	0.90 m	3.59 m		
fter optimisation	0.64 m	0.67 m		

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GSAT0201/0202 GREAT Experiment

- **Opportunity for Scientific Community**
 - use the satellites to test the laws of General Relativity and fundamental physics
 - based on the excellent stability of the PHM clocks in a varying gravitational environment *
- Features supporting GREAT (Galileo gravitational Redshift Experiment with eccentric sATellites)
 - High eccentricity of the final orbits of 2 satellites of approx. 0.156 *
 - Highly stable on-board time reference via Passive Hydrogen Maser (PHM) space clocks *
 - Design lifetime allows for long-term observations *
 - Equipped with laser retro-reflector to support satellite laser ranging *
 - Satellites permanently monitored by IGS \star

Objective of the Experiment

- Orbit eccentricity induces periodic modulation of gravitational redshift at orbital frequency *
- Highly stable clocks allow to monitor this effect by observing periodic change of clock rate *
- Change of clock rate is related to the periodic variation of the gravitational potential. \star
- Averaging these measurement over many orbits (≥1 year) will increase measurement * accuracy and allow to push the current state of art by about 1 order of magnitude.
- Today State of the Art: * Gravity Probe-A - gravitational redshift verified with an accuracy of 1.4 x10-4 in 1976

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GSAT0201/0202 GREAT Experiment



• Who is Involved

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Two parallel contracts launched (Oct 2015) by ESA with SYRTE/Observatoire de Paris and ZARM/University of Bremen to perform these tests

European Space Agency

Systèmes de Référence Temps Espace

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The tests are encouraging and consolidated results are expected by Oct 17. ESA UNCLASSIFIED - For Official Use

GSAT0104 (E20) IOV Satellite Usage



- Since May 2014 **GSAT0104 (E20) only transmitting E1 signal** with reduced power
- Galileo is a dual-frequency system but a high proportion of civil users will foreseeably use single frequency receivers, therefore a recovery action has been undertaken to allow use of GSAT0104 for user positioning as well as SAR applications
- First Step Prototyping: to define and prototype (in TGVF) a ground processing method to achieve this, using the code&phase GRAPHIC observables in place of dual frequency IONO-free combinations
 - The navigation message generated is consistent with the message broadcast by the nominal constellation (e.g. Clocks and Broadcast Group Delays) allowing seamless combination at receiver level PVT processing
- Second Step Galileo System Ground Mission Segment:
 - Definition of implementation currently under finalisation for inclusion in operations



GSAT0104 (E20) Performance



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GTRF17v02 comparison of to **ITRF2014**



Transformation parameters from GTRF17v02 to IGS14 (ITRF2014) indicate a few millimeter (and mm/yr) level agreement between the two frames.

	T1 mm	T2 mm	тЗ mm	D 10-	-9	R1 mas	R2 ma	S	R3 mas	Epoch Y
	0.0	0.0	0.0	0.0	00	0.000	0.	000	0.000	17:001
+/-	0.2	0.2	0.2	0.0	03	0.007	0.	008	0.008	
Rates	0.0	0.0	0.0	0.0	00	0.000	0.	000	0.000	
+/-	0.2	0.2	0.2	0.0	03	0.007	0.	008	0.008	
	Station #	RMS	-Pos.		Epc	och	RMS	-Vel.		
		E	N	U			E	N	U	
			mm		У	7		mm/y		
	127	2.5	2.0	4.0	17:	001	0.4	0.3	0.6	
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Metadata publication request

Requested by Scientific community

• Galileo scientific advisory committee (GSAC)

• International GNSS Service (IGS)

IGS	
T	20. August 2012
International GNSS Service Formerly the International GPS Service IGS Coverning Roard Newborc Us Hagenothe, Consumy Childrane Dation Algorithm, France	Mr. Paul Flarment Galileo Programme Manager European Commission Enterprises and Industry Directorate Rue de Mot, 28 B-1049 Brussels
E. Felicitas Arlas, France BIPMCCTF Representative Claude Brocher, France	cc: Gerhard Beutler, Bertram Arbesser-Rasiburg
IEEE Representative Carine Brepalan, Bulgians Math Coines, Conside	Dear Mr. Flament.
Yanin Dang, China Shalino Dunai, USA Banino Dunai, USA Banino Garajin, Franco HGB Relowner France Coordinator Halo Gardina, USA Andraisti Cantor Coordinator Christian Rechtanon, USA Gary Memoto, Anatralia	Prof. Gerhard Bestler, Chair of ESA's Galileo Science Advisory. Committee (GSAC), requested in his letter of October 13, 2011 access to meta data related to the Galileo satellites and sensor nations for the scientific community. On behalf of the International GNSS Service (IGS) I strongly support this request and would like to add a few remarks from the IGS prespective.
Kon Mull-and, Canada Coush Meettinas, USA Oliver Monambrock, Germany Bach E. Nolas, USA Director, NGT Control Revisio Corey Soli, USA James Perk, South Karisa Chris Kano, Assanzia MGD President Jacob Potenter, Contany	The information and data requested by Prof. Beutler are essential for the precise modeling of observations and orbits of the Galileo statellines, for the consistent and interoperable integration of Galileo products into the highest quality produces of the IGS, and for a consistent alignment of terrestrial reference frames. Our products serve Earth system research, hazards detection, and many more applications relevant to society as a whole.
Stotas Schare, Switzertlad Ball Schnit, Gennany Talo Schare, Gennany Kolson, USA Clovic Product Continuum Echard Wonnacer, South Alvica March Zeibert, Linite Kingdon	The list provided by Prof. Beatler represents in essence a complete list of the required information. From the IGS perspective the following subset of information is needed urgently in the nearest future. This information is crucial for a correct high precision modeling of the Galileo observables, in particular for the consistent determination of satellite clock corrections:
Prof. Dr. Urs Hugenstehler hattoner for Astronomical ered Physical Geodesy Techniche Geoerstehlt Bläncken Arstantung 21 3-30331 Montrin, Germany	 nereignation antenna phase center offsets in the body-fixed reference frames w.r.t. the satelliteric centers of mass for each carrier frequency and for all Galileo satellites (together with the definition of the body- fixed reference frames).
Tel: 4949-205-2019) Fai: 49-39-205-2019 E-mil: withoposible/Olivity-manufactor.de	 aminule models for the Galileo satellites (which are of particular importance, if the antenna phase center is not located on the axis connecting the satellite's center of mass with the center of mass of the Earth).
here the over	The IGS would be most obliged, if the above information could be made available to the scientific community with highest priority.
The second secon	Sincerely yours

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Metadata content

Requested by

- Galileo scientific advisory committee (GSAC)
- International GNSS Service (IGS)

Status

- Galileo IOV Satellite Metadata released during Initial Service Declaration.
- Galileo FOC metadata to be released.

Content

- Attitude Law
- Mass and Centre Of Mass evolution
- Navigation Antenna Phase Centre Corrections
- Geometry and optical properties
- Laser Retro Reflector Location
- Satellite Group Delay







Metadata location



https://www.gsc-europa.eu/support-to-developers/galileo-iov-satellite-metadata#2 https://ilrs.cddis.eosdis.nasa.gov/missions/satellite_missions/current_missions/ga01_com.html





Ground Segment Upgrades

Ground Control Segment 2.1.2

- Deployed at GCC-D and GCC-I
- Spacecraft control automation
- 26 satellite capability
- Business Continuity
- Additional TTCF-6 in Papeete

Ground Mission Segment 2.2

- Deployed at GCC-D and GCC-I
- GSAT201/202 support
- Seamless PTF switch capability
- Business Continuity
- Additional GSS and ULS redundancy

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FOC Satellites Production Status

- FOC-M7 (L9 Ariane-5 end 2017) \star
 - **FM15** in OHB Bremen
 - \star FM16 in OHB Bremen
 - 🔸 FM17 in OHB Bremen
 - **FM18** in OHB Bremen

FOC-M8 (L10 Ariane-5 mid 2018)

- FM19 in OHB Bremen
- **FM20** in OHB Bremen
- **FM21** in ESA ESTEC, under testing
- **FM22** in OHB Bremen, under testing







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BATCH3 Satellites Procurement



- * Batch 3' satellites Contract to build and test another 8 satellites signed at the Paris Air and Space Show on 22 June 2017
- Awarded to a consortium led by prime contractor OHB with Surrey Satellite Technology Ltd as payload manufacturer



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Launch #9 (Ariane 5) – December 2017





GSAT0215 - GSAT0216 GSAT0217 - GSAT0218 SAT 19-20-21-22 Plane A slots 1,3,4,7





Increased Availability of Positioning



L8: 4/4/7



79.6%

86.0%

99.8%



Horizontal Dilution Of Precision

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4/4/7

4/4/7+2

8/4/7+2

L8

L8 + L3

L9

European Space Agency

74 / 2.58 hrs

62 / 3.33 hrs 4 / 59.89 hrs

(~8 min average time with insufficient geometry)

GNSS Evolutions activities



- European GNSS Evolution Programme started in 2008 Phasing out in 2018
- Continuity of R&D ensured through H2020 already since 2016
- EGEP invested 170 M€ (70+ contracts) in GNSS R&D:
 - EGNOS V3 and Galileo G2 Definition Phase.
 - Galileo Evolutions Technology pre-developments
- ESA •
- Science and GNSS Transversal R&D.



Galileo Evolution Scenarios



Mission Evolution Scenario ES-1 ("Basic")

Galileo FOC optimisation scenario in terms of performance and operability:

- similar spacecraft platform envelope with the same Signal In Space
- improved performance
- limited set of new services
- technology obsolescence

Mission Evolution Scenario ES-2 ("Medium")

a first step in the alignment of Galileo with other GNSS Systems competitiveness.

- medium sized platform
- new signals/services
- increased robustness & capabilities
- enhanced performance

Mission Evolution Scenario ES-3 ("Ambitious")

the state of the art

- medium/large size platform
- advanced features
- flexibility of service provision
- high performance and robustness

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Galileo Evolutions Scenarios





Conclusions



- → Initial Services Declaration on 15th December 2016
- → 18 satellites in orbit and production of remaining 8 planned to be completed in 2017
- Constellation deployment "boosted" by Ariane-5 launch capability fully qualified: next Launch#9 in December `17
- Procurement of additional 3rd Batch of 8 satellites initiated
- → Galileo Core Infrastructure handed over to EC-GSA to start Exploitation Phase (Galileo Service Operator)
- Ground Segment deployment continues to support stable and continuous availability of Galileo Signal-in-Space to users
- → GNSS Evolution R&D well under way

