Joint Reprocessing of GPS, GLONASS and SLR Observations - First Results

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Reprocessing





Observation data and modelling:

GNSS Orbit Overlaps



Satellite Clocks

Station and satellite clock computation steps: 1. Generally, processing includes 24:00:00 UTC observation epoch 2. Pre-process and clean zero-difference code observations 3. Compute inter-system and inter-frequency code biases 4. Pre-process and clean zero-difference phase observations 5. Introduce and fix final 3-day orbits and Earth rotation parameters 6. Based on ionosphere-free linear combination, estimate station coordinates, troposphere and 5-min station and satellite clocks 7. Interpolate 5-min clocks to generate 30-sec clocks



- In total, 340 GNSS stations with 140 providing GLONASS and 70 SLR stations (starting on 1st Jan., 1994), see Fig. 1

- Consistent modelling of microwave and laser observations

- Starting with 1st Jan., 2002, rigorous GPS/GLONASS combination - Major modelling aspects

terrestrial reference frame : ITRF2008/IGS08 *Earth tides* : IERS2010 conventions ocean tide loading : multi-mission altimetry model EOT11a *atmospheric tidal loading* : S_1+S_2 tides (Ray and Ponte, 2003) *atmospheric non-tidal loading* : GRACE AOD1B de-aliasing products antenna phase center : IGS08.atx *troposphere modelling* : 6 hour VMF1 grids



Fig. 2: Number of reprocessed stations and satellites. GLONASS is processed starting with January 1, 2002. Top: total number of GNSS-tracking receivers (black), number of GPS+GLONASS-tracking receivers (grey) and number of SLR stations. Bottom: total number of GPS (black), GLONASS (grey) and block-specific number of satellites.

Terrestrial Reference Frame

Combining daily solutions from 1 January 2002 to 31 De-

Fig. 4: Comparison of daily mean orbit overlaps. Top: GPS satellites from GPSonly and GPS+GLONASS combined solutions. Bottom: GLONASS satellites from GLONASS-only and GPS+GLONASS combined solutions. Eclipsing satellites and satellites with extraordinary large overlaps are omitted.

Fig. 4 shows the impact of combining GPS and GLONASS on the estimated satellite orbit quality. In general, no significant improvements can be obtained for the GPS satellite orbit overlaps. However, the quality of the GLONASS orbits significantly improves when computed from a GPS/GLONASS combined solution.

GNSS Orbit Validation Using SLR







Fig. 6: Standard deviations (STD) of 30-sec satellite clocks (offset and drift removed) from GPS (top panel) and GLONASS (bottom panel). At the right: Distribution of the satellite clocks according to the STD.

Separately for GPS and GLONASS, Fig. 6 shows daily standard deviations (STD) of the 30-sec satellite clocks for the year 2008. Since the clocks can be connected at the midnight epoch, it is possible to investigate their longterm behaviour. Accordingly, Fig. 7 exemplarily shows the Allan deviation computed over a 10 day interval.

cember 2011, GNSS-only terrestrial reference frames are derived. Minimum constraint solutions are generated for station coordinates and velocities applying no-net rotation conditions w.r.t. IGS08. Translation rates and scale rate (Tab. 1) have to be referred to the ITRF2008 where the origin is realized by SLR and the scale by VLBI. Including GLONASS doesn't cause a loss of precision of the results.

Tab. 1: Terrestrial reference frame (TRF) comparisons. Results of a 14-parameter similarity transformation of IGS08 w.r.t. individual GNSS-only TRFs (rotations are defined to be zero, hence, omitted here).

IGS08 w.r.t.	Translations [mm]/ Translation Rates [mm/y]			Scale [mm]/ Scale Rate [mm/y]
	Х	Ŷ	Z	
GPS+GLONASS	-4.1 ± 0.1 - -1.0 ± 0.0 -	-6.5 ± 0.1 +1.3 ± 0.0	$-3.7 \pm 0.1 +0.5 \pm 0.0$	-1.9 ± 0.2 0.0 ± 0.0
GPS-only	-4.3 ± 0.1 - -1.1 ± 0.0 -	-6.7 ± 0.1 +1.4 ± 0.0	$-3.9 \pm 0.1 +0.5 \pm 0.0$	-1.8 ± 0.2 0.0 ± 0.0

Earth Rotation Parameters



Fig. 5: SLR residuals (color bar in mm) for selected GPS (top) and GLONASS (bottom) satellites in a Sun-fixed reference frame (elevation of Sun above the orbital plane β_0 , angle Δu in the orbital plane).

Applying a homogeneous and consistent processing to both laser and microwave observations, the mean biases (Fig. 5) could be reduced for GPS from about 36 mm (Flohrer, 2008) to about 11 mm (Tab. 3). Compared to GPS, a larger mean RMS of 33 mm for GLONASS still indicates difficulties associated with the orbit modeling of these satellites. However, the estimated mean biases are smallest for GLONASS.



Fig. 7: Allan deviation (ADEV) of GPS and GLONASS 30-sec satellite clocks. The ADEV was computed from one interval of 10 days for each satellite where only few data was missing.

Fig. 3: Earth rotation parameter estimates w.r.t. the official IGS series for different types of solutions: GPS-only, GLONASS-only and GPS/GLONASS combined.

Tab. 2: Standard deviations of the Earth rotation parameter estimates w.r.t. the official IGS series for different types of solutions: GPS-only, GLONASS-only and GPS/GLONASS combined.

	Glonass-only	GPS-only	GPS/GLONASS
X _P 10 ⁻³ ["]	0.3559	0.0800	0.0747
Y_{P} 10 ⁻³ ["]	0.3381	0.0886	0.0858
\dot{X}_{P} 10 ⁻³ ["/d]	5.5304	2.4540	2.2667
\dot{Y}_{P} 10 ⁻³ ["/d]	6.4855	2.5222	2.4148
LOD [ms/day]	0.2623	0.1277	0.1154

Tab. 3: Mean and Root Mean Square (RMS) of SLR residuals w.r.t. GNSS orbits. Satellite System/ Orbital Number of SLR Mean residual RMS of residuals Satellite Type palne [mm] observations [mm] GLONASS 257029 -1.9 33.2 all planes 102200 GPS all planes 20.8 -11.4 GLONASS plane 1 113914 -2.7 31.4 33.3 GLONASS plane 2 70316 -2.4 GLONASS 70157 plane 3 -0.1 35.5 GLONASS 74283 38.0 all planes -0.8 GLONASS-M all planes 182900 -1.9 33.0 GLONASS-K all planes 2642 -6.6 32.2

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