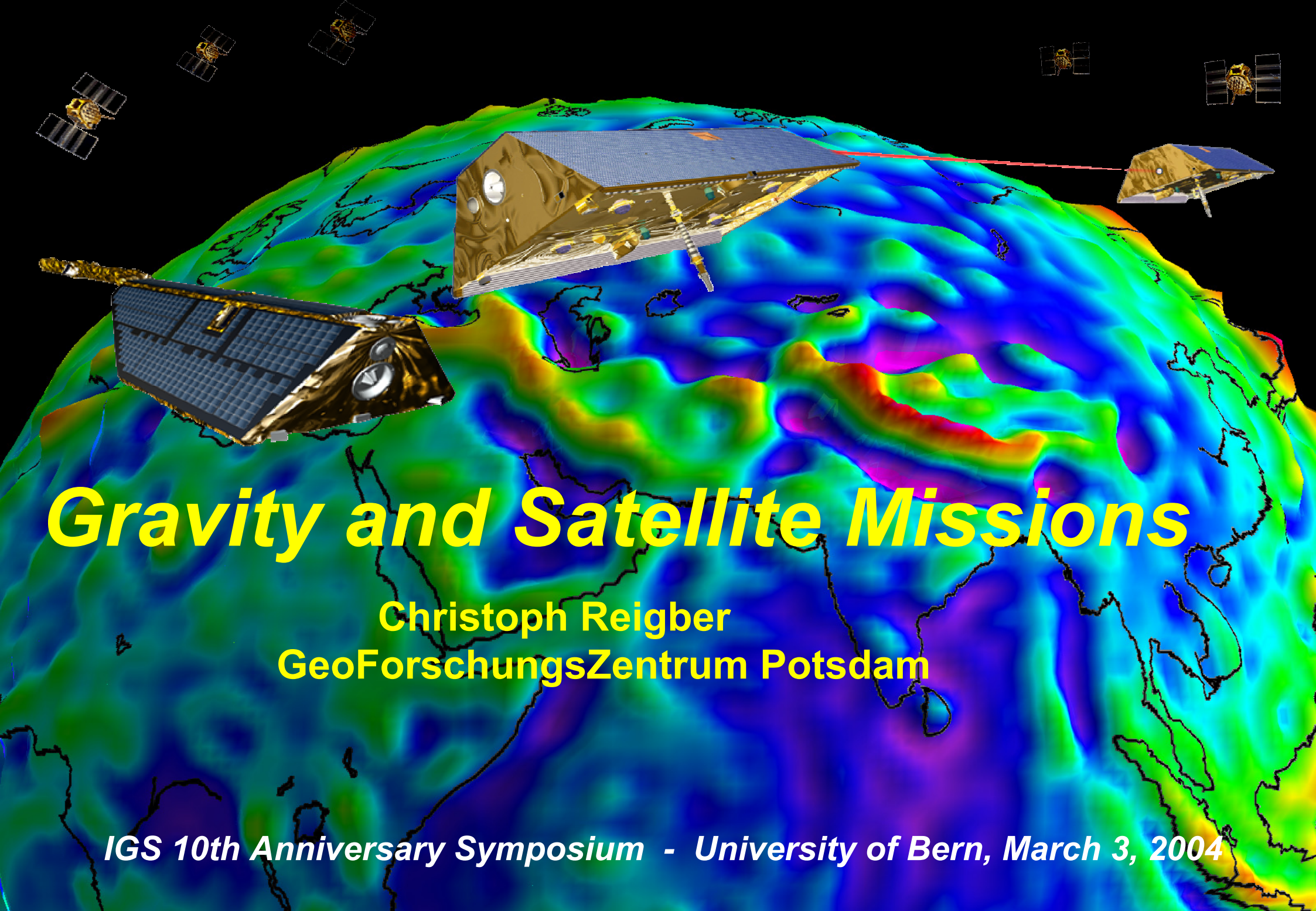




Gravity and Satellite Missions

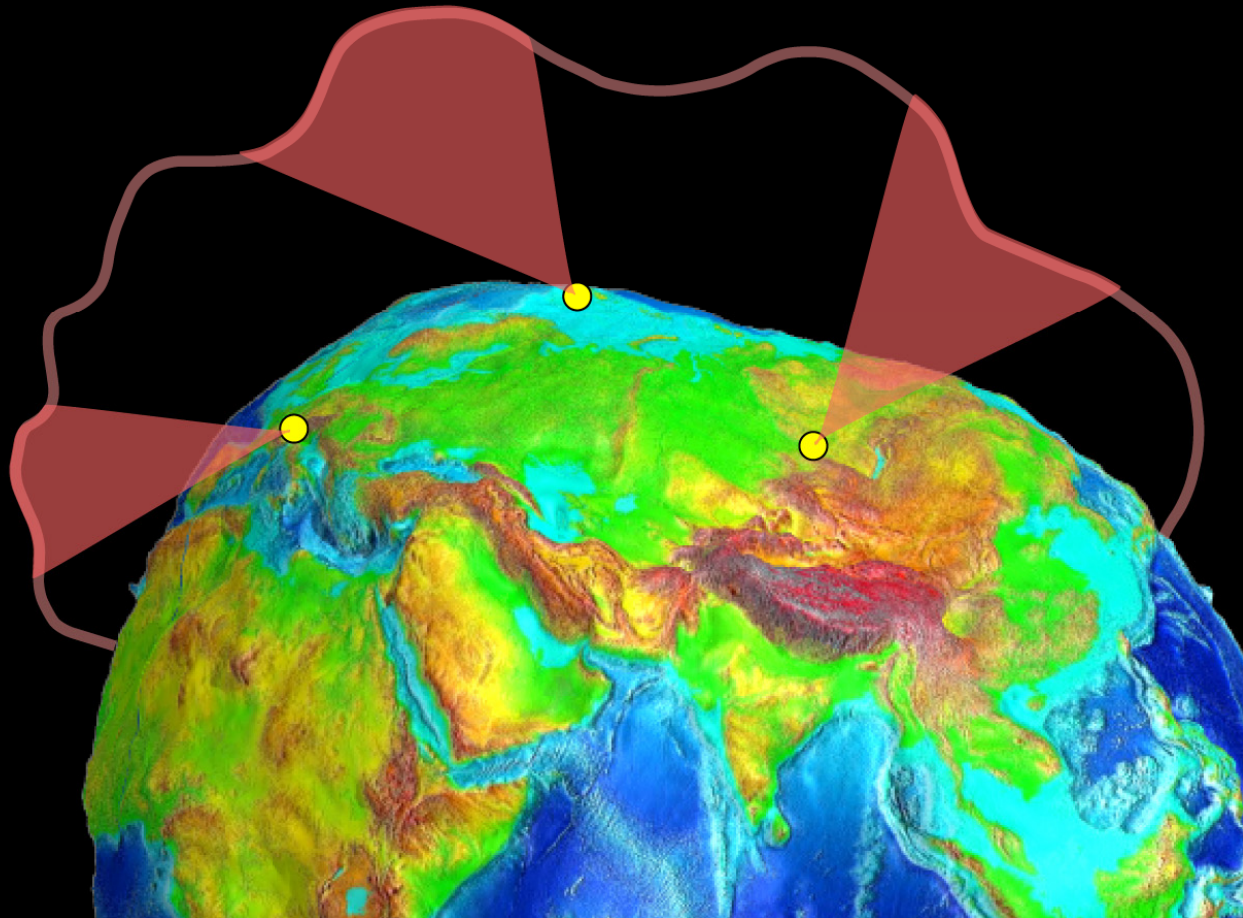
**Christoph Reigber
GeoForschungsZentrum Potsdam**

IGS 10th Anniversary Symposium - University of Bern, March 3, 2004

Typical Ground-Based Coverage

(before GPS Space Receiver became available)

- One dimension at a time
- Large coverage gaps
- Very critical for LEOs



DOI Process for Gravity Field Recovery

$$\Delta r_i^T = [\Delta x_i, \Delta y_i, \Delta z_i] = f(\Delta C_{lm}, \Delta S_{lm})$$

Pseudo-observations

$$A = \left\{ \frac{\partial (\Delta x_i, \Delta y_i, \Delta z_i)}{\partial (\Delta C_{lm}, \Delta S_{lm})} \right\}$$

Design matrix from partials

$$\{\Delta x_i, \Delta y_i, \Delta z_i\}$$

Observation residuals

LSA

$$\{\Delta \hat{C}_{lm}, \Delta \hat{S}_{lm}\}$$

Res. harmonic coeff.

Gravity Field Representation & Functionals

$$T = \frac{GM}{R} \sum_{l=2}^L \sum_{m=0}^l \left(\frac{R}{r} \right)^{l+1} P_{lm}(\cos \vartheta) \left[\Delta \hat{C}_{lm} \cos m\lambda + \Delta \hat{S}_{lm} \sin m\lambda \right]$$

RESIDUAL
GRAVITATIONAL
POTENTIAL
in m²/s²

100 km resolution requires $L = 20000 \text{ km} / 100 \text{ km} = 200$

RESOLUTION

$$N = R \sum_{l=2}^L \sum_{m=0}^l P_{lm}(\cos \vartheta) \left[\Delta \hat{C}_{lm} \cos m\lambda + \Delta \hat{S}_{lm} \sin m\lambda \right]$$

GEOID HEIGHTS
in meter

$$\Delta g = \gamma \sum_{l=2}^L (l-1) \sum_{m=0}^l P_{lm}(\cos \vartheta) \left[\Delta \hat{C}_{lm} \cos m\lambda + \Delta \hat{S}_{lm} \sin m\lambda \right]$$

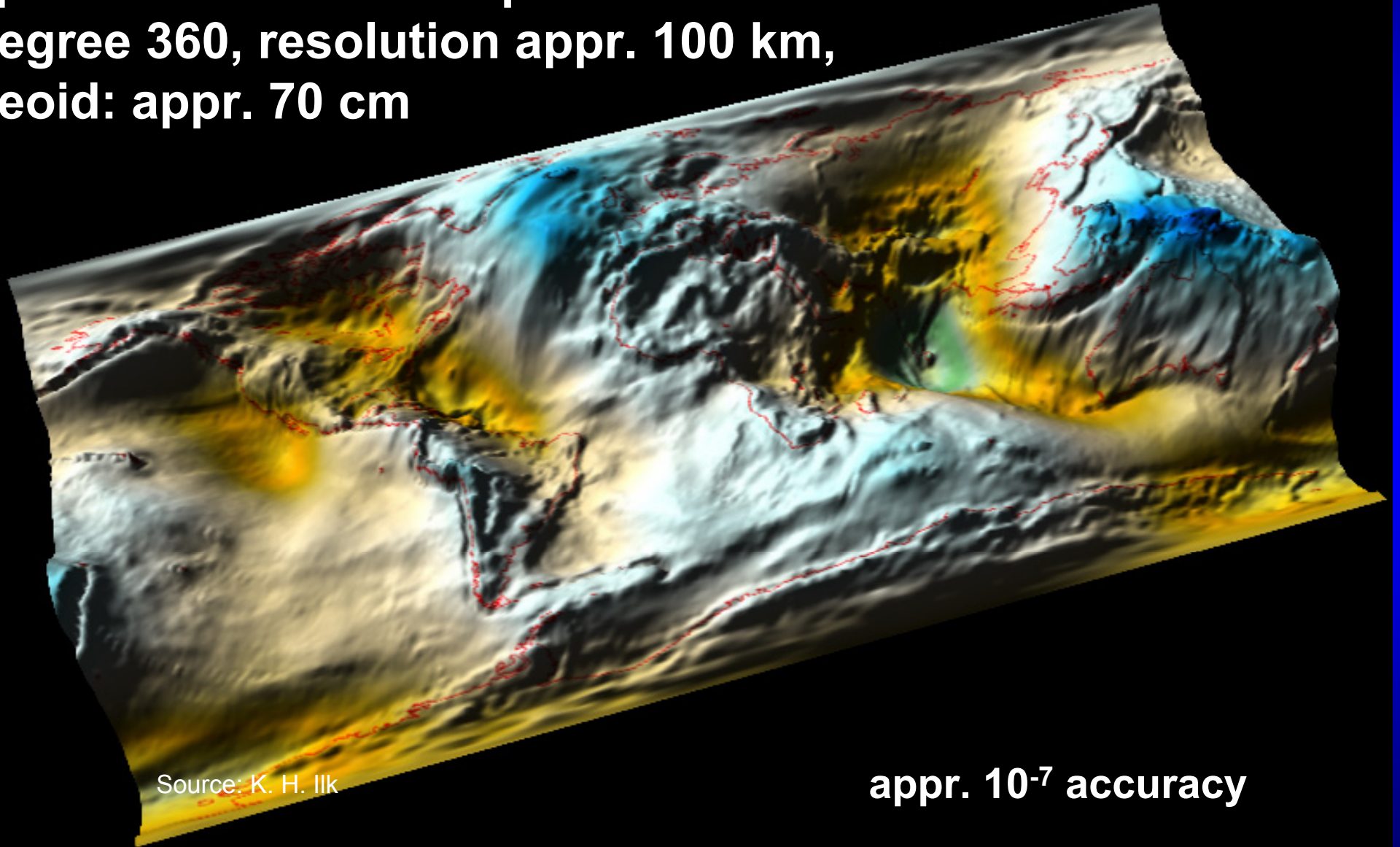
FREE AIR GRAVITY
ANOMALIES
in mgal (10⁻⁵ m/s²)

$$c_l = \sum_{m=0}^l \left[C_{lm}^2 + S_{lm}^2 \right]$$

Power Spectral Density (PSD)
Average Signal Strength

Earth Gravity Field in 2000

spherical harmonics up to
degree 360, resolution appr. 100 km,
geoid: appr. 70 cm



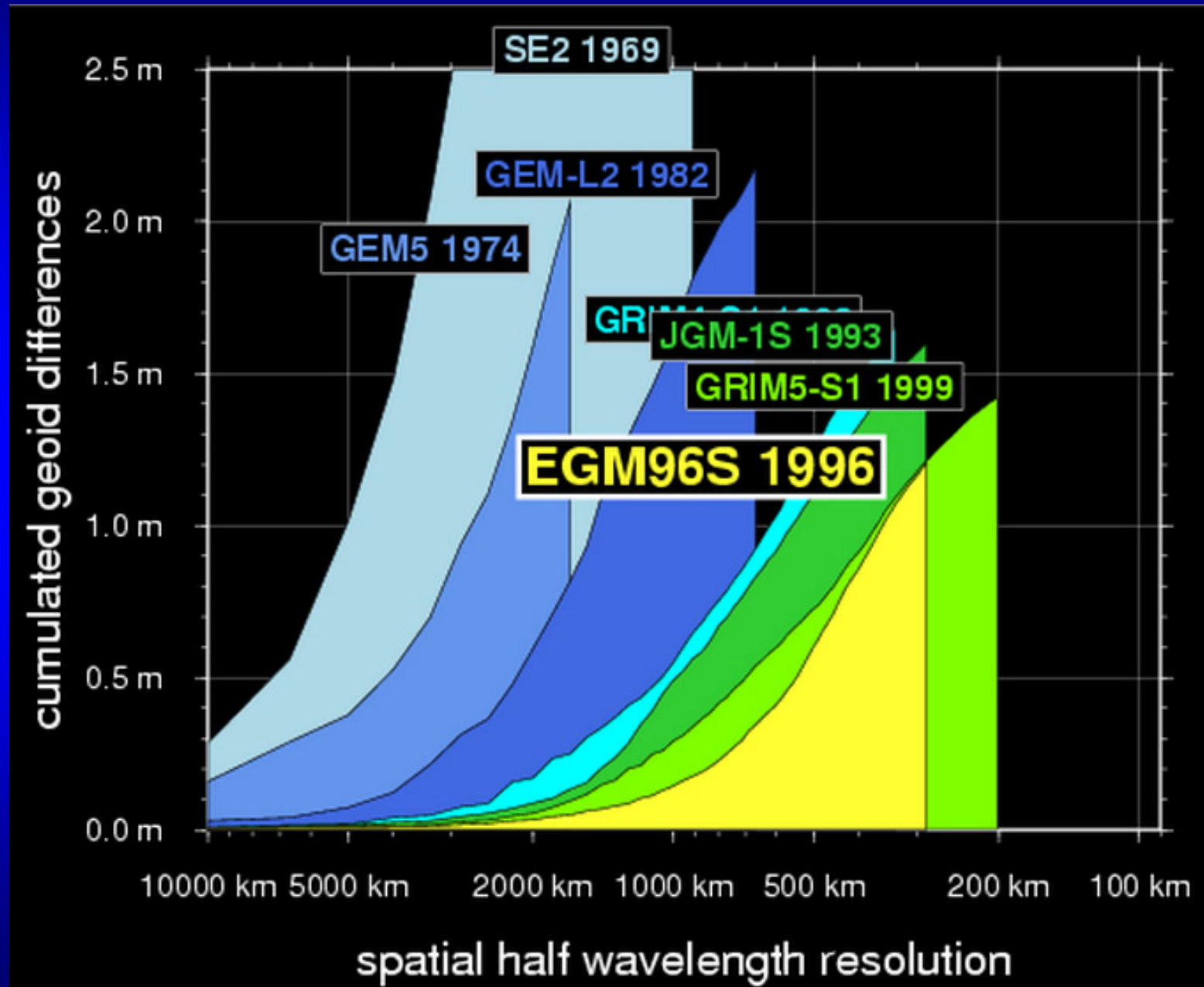
Source: K. H. Ilk

appr. 10^{-7} accuracy

Pre-CHAMP/GRACE Satellite Gravity Field Models

Model	Institution	Period
<i>SE I-III</i>	Smithonian Astrophysical Observatory	1966 - 1973
<i>GEM 1-10/T1-3</i> <i>EGM 96</i>	Goddard Space Flight Center/ Ohio State University	1972 - 1998
<i>GRIM 1-5</i>	DGFI/GFZ Potsdam/GRGS Toulouse	1976 - 2000
<i>TEG 1-4</i>	CSR/UTEX Austin	1988 - 2001

Pre-CHAMP/GRACE Satellite Gravity Field Models



New Observation Techniques

Satellite-to-Satellite Tracking

◆ *high- low*

Wolff, 1969

◆ *low- low*

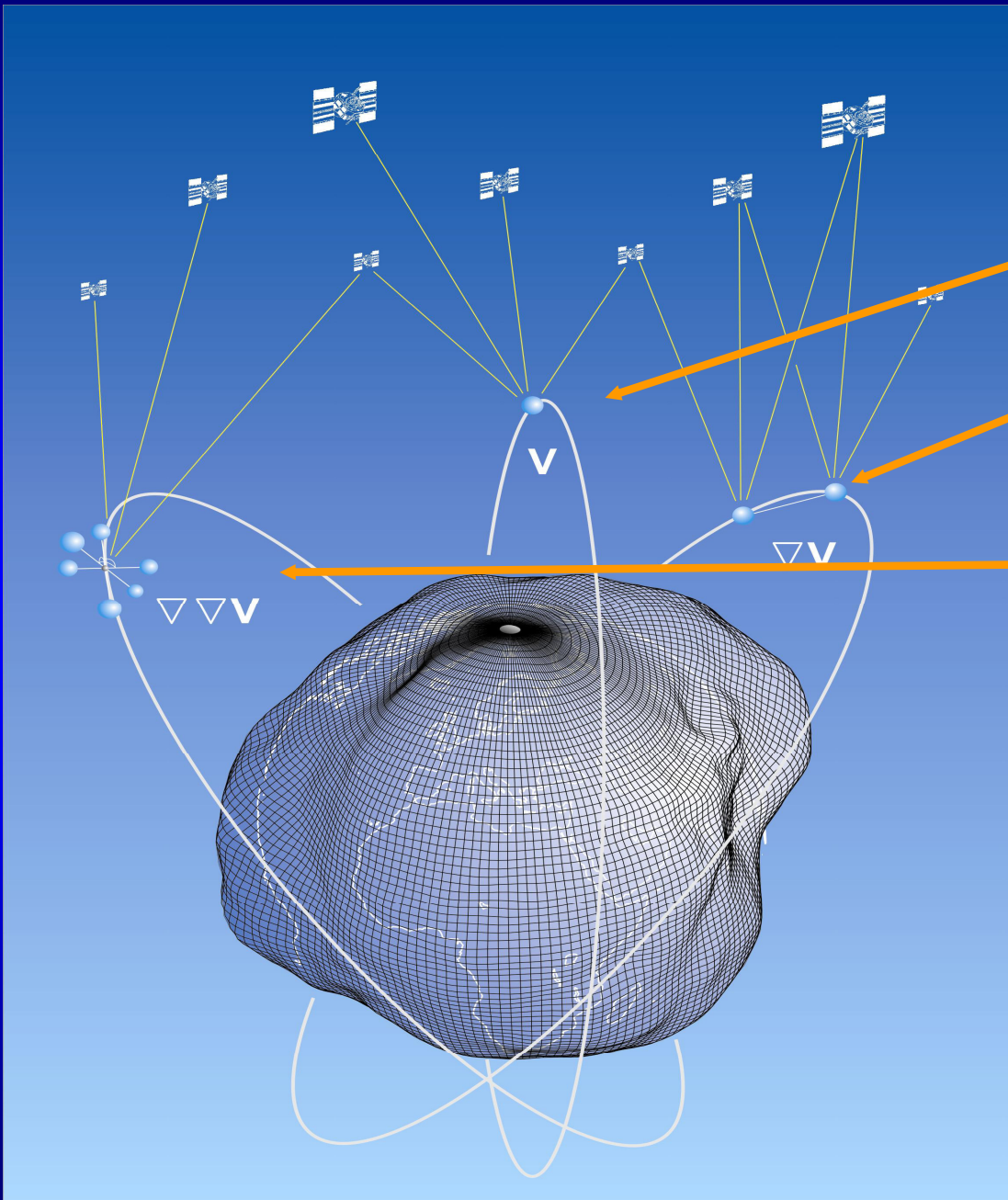
Satellite Gravity Gradiometry

Williamstown 1969, NASA

Elmau 1979, ESA

Erice 1985, IAG

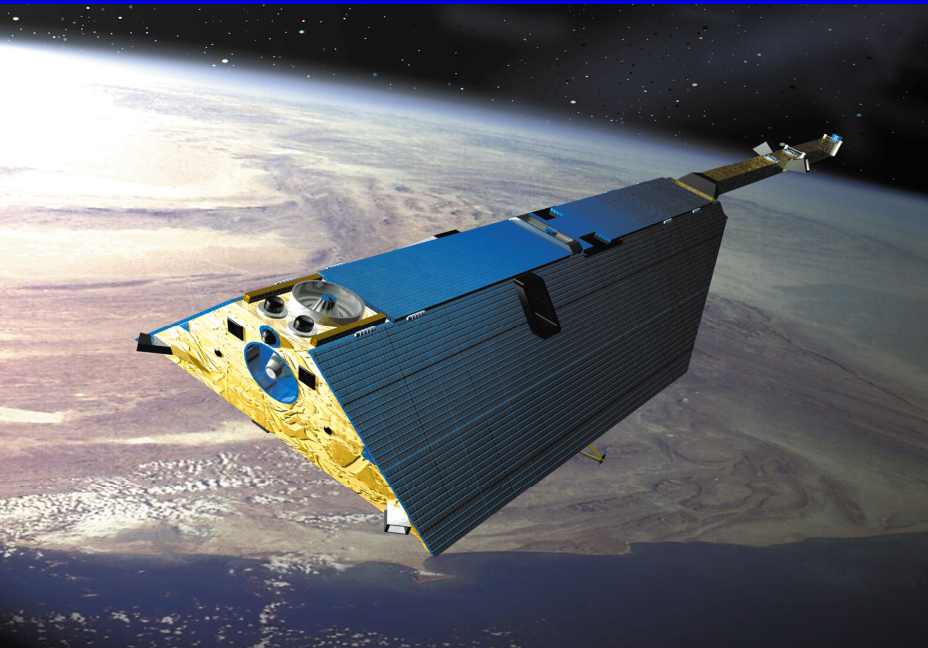
Coolfont 1991, NASA



The Dawn of a New Age

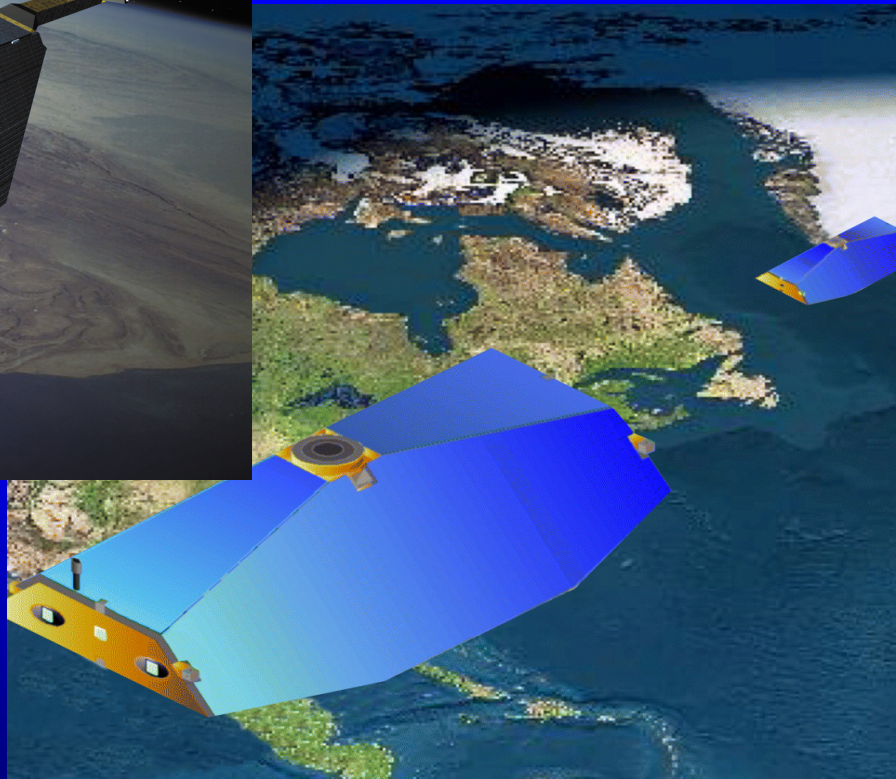
CHAMP (2000)

Germany



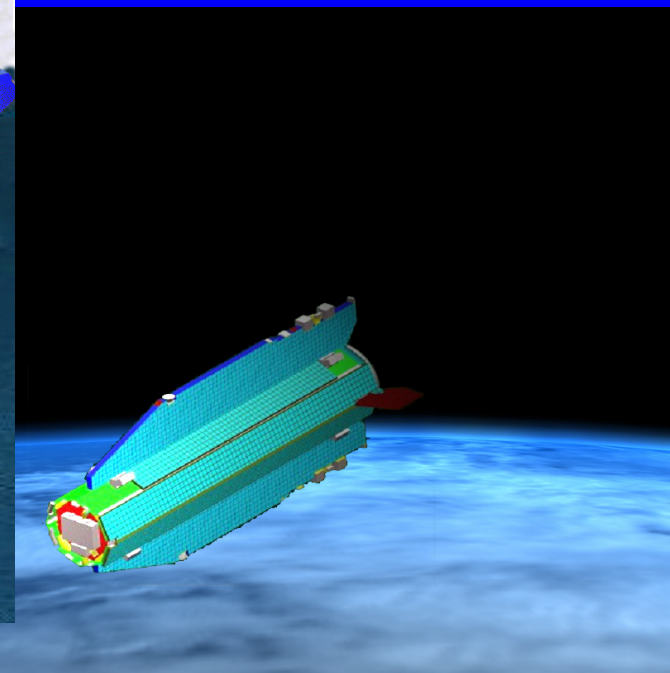
GRACE (2002)

USA-Germany



GOCE (2007)

European



CHAMP

Mission of GFZ Potsdam & DLR (Germany)
with contributions of NASA (USA), CNES (France) and AFRL (USA)

Launch: Juli 2000 **Mission duration:** 5 yrs

Orbit: nearly circular polar orbit (87°), 454 km altitude

PI: Christoph Reigber / GFZ



Mission goals:

- ◆ Gravity field mapping: medium resolution static & time-variable components
- ◆ Magnetic field mapping: main & crustal field and time-variable components
- ◆ Atmosphere & ionosphere sounding

Payload:

- ◆ 3-D accelerometer ($3 \times 10^{-9} \text{ m/s}^2$).
- ◆ 16-channel GPS receiver (high-low SST)
- ◆ Laser-retro-reflector
- ◆ Magnetometer
- ◆ Ion drift meter
- ◆ Star sensors

GRACE

Mission of NASA (USA), DLR (Germany), JPL(USA) , CSR/UTEX(USA) & GFZ Potsdam (Germany) within NASA's ESSP Programme

Launch: Mar 2002 **Mission duration:** 5 yrs

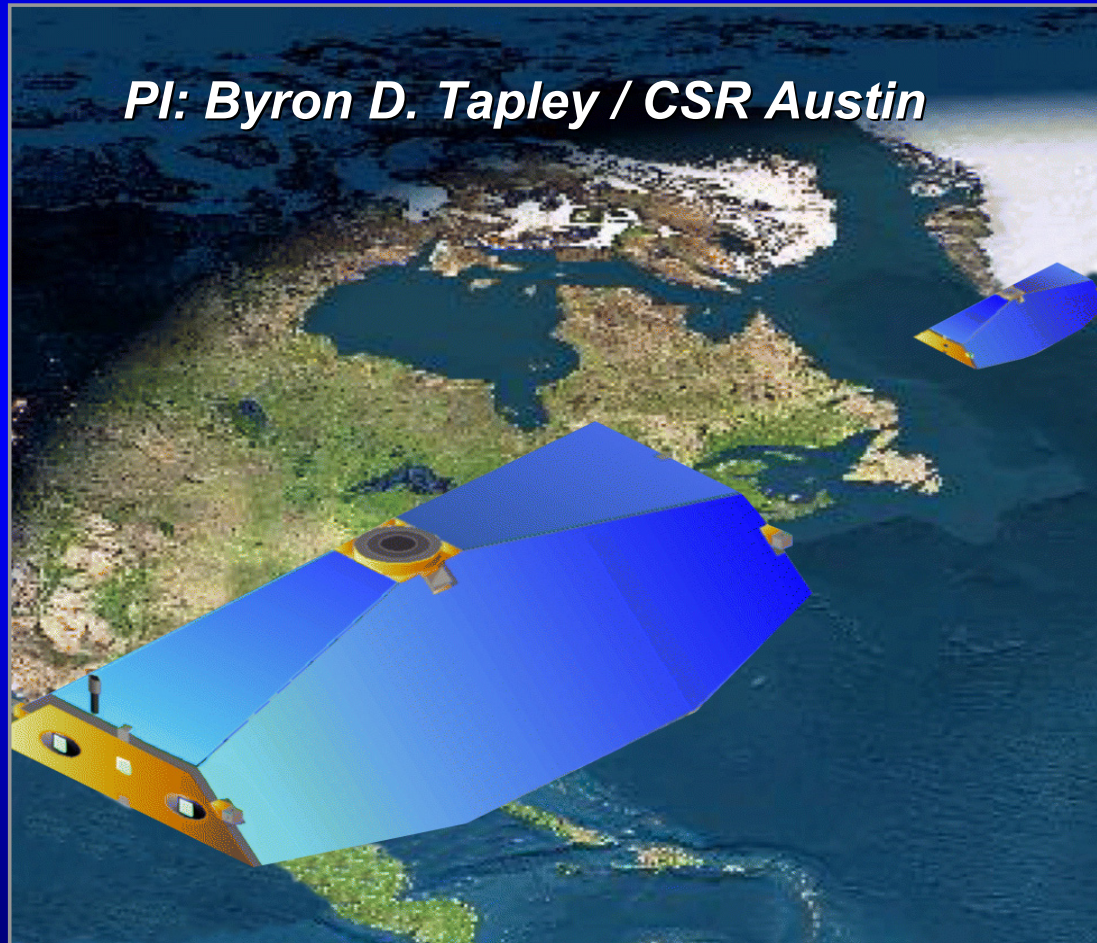
Orbit: nearly circular polar orbit (89°), 500 km altitude

Mission goals:

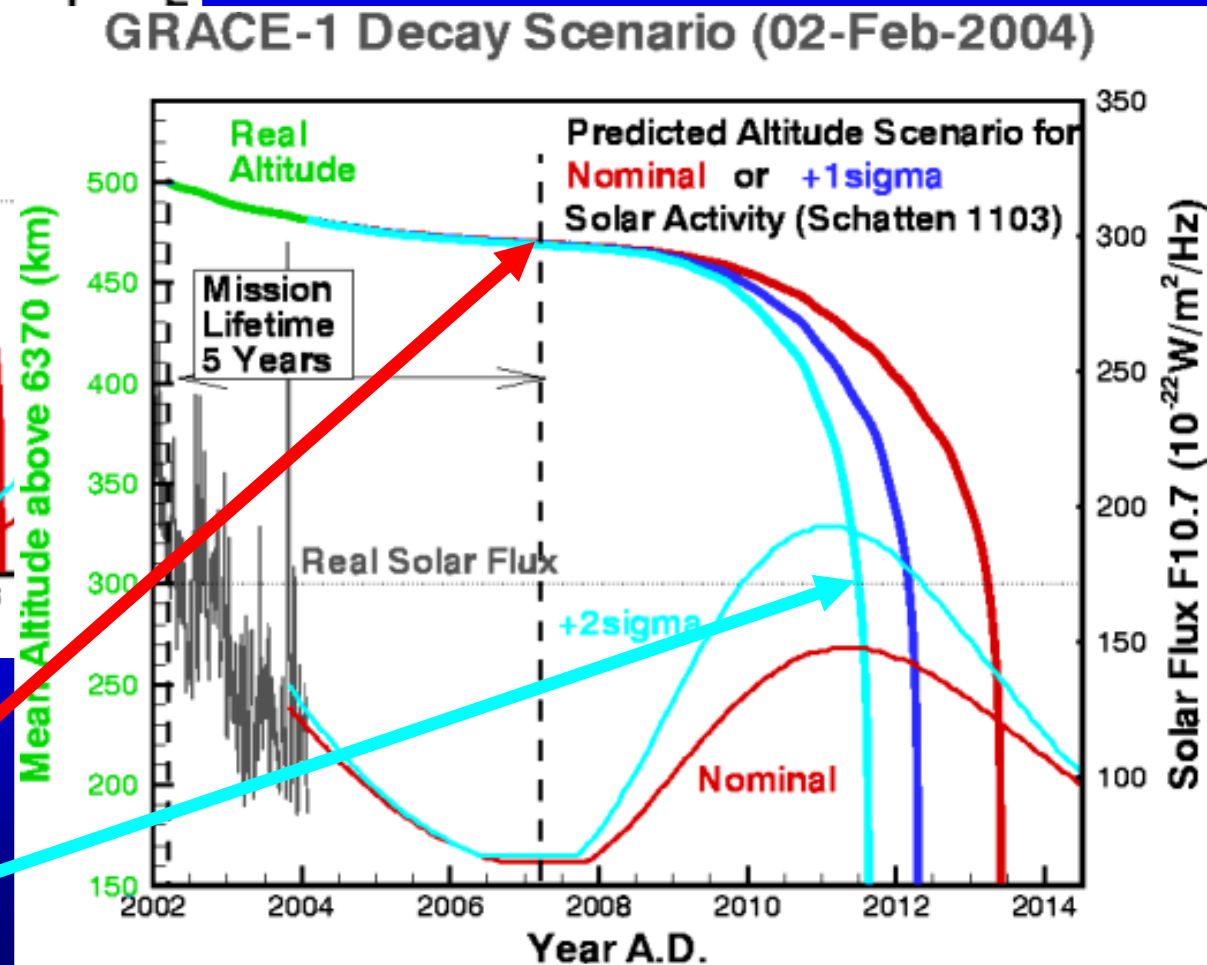
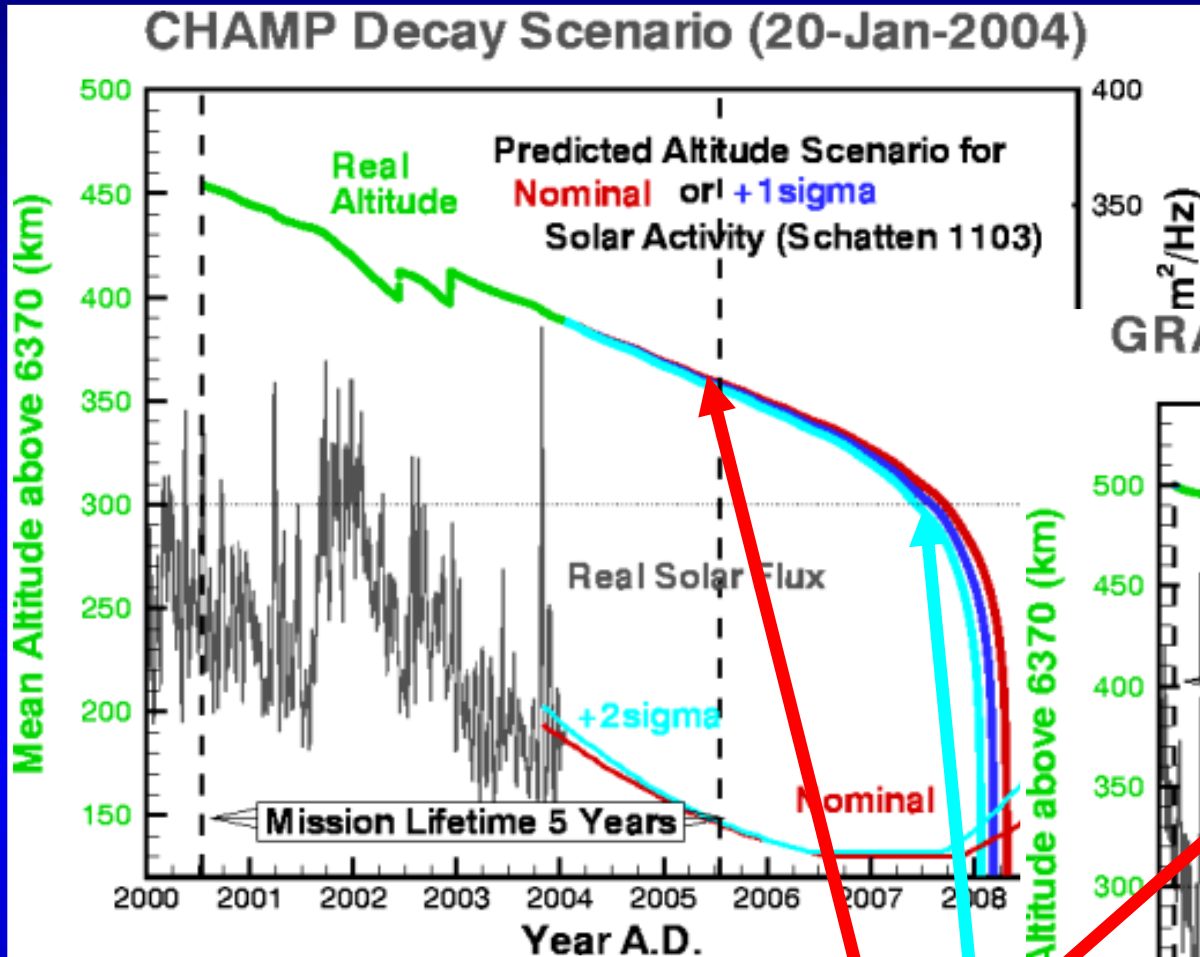
- ◆ High resolution, long- and medium-wave, mean & time- variable gravity field mapping
- ◆ Atmosphere sounding

Payload:

- ◆ Microwave distance ranging between the satellites ($<5 \mu$), relative velocity ($<0.5 \mu/s$) (low-low SST)
- ◆ 3-D accelerometer (10^{-10} m/s^2)
- ◆ 24-channel GPS receiver (high-low SST)
- ◆ Laser-retro-reflector
- ◆ Star sensors



CHAMP & GRACE Orbit Altitude Evolution



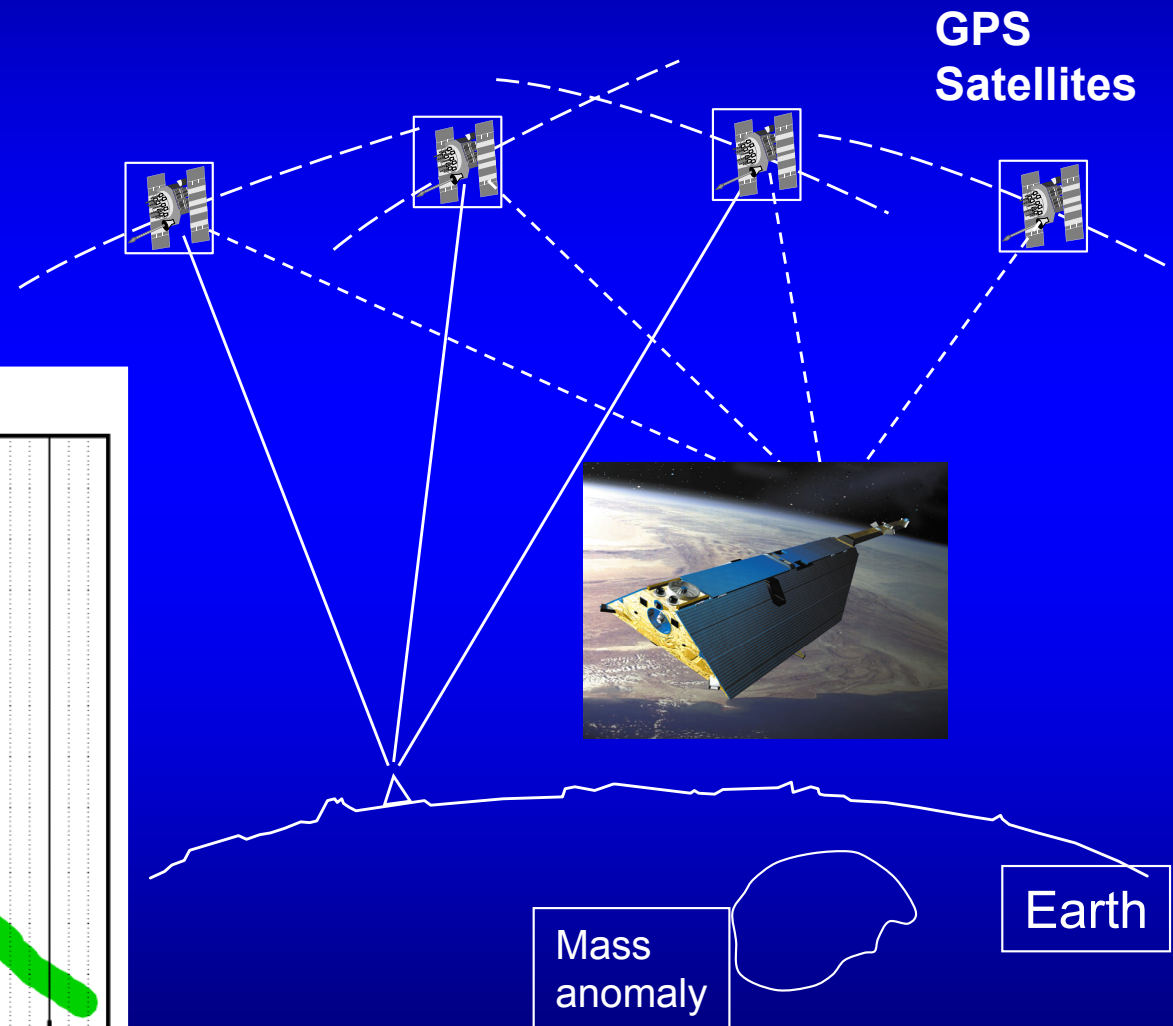
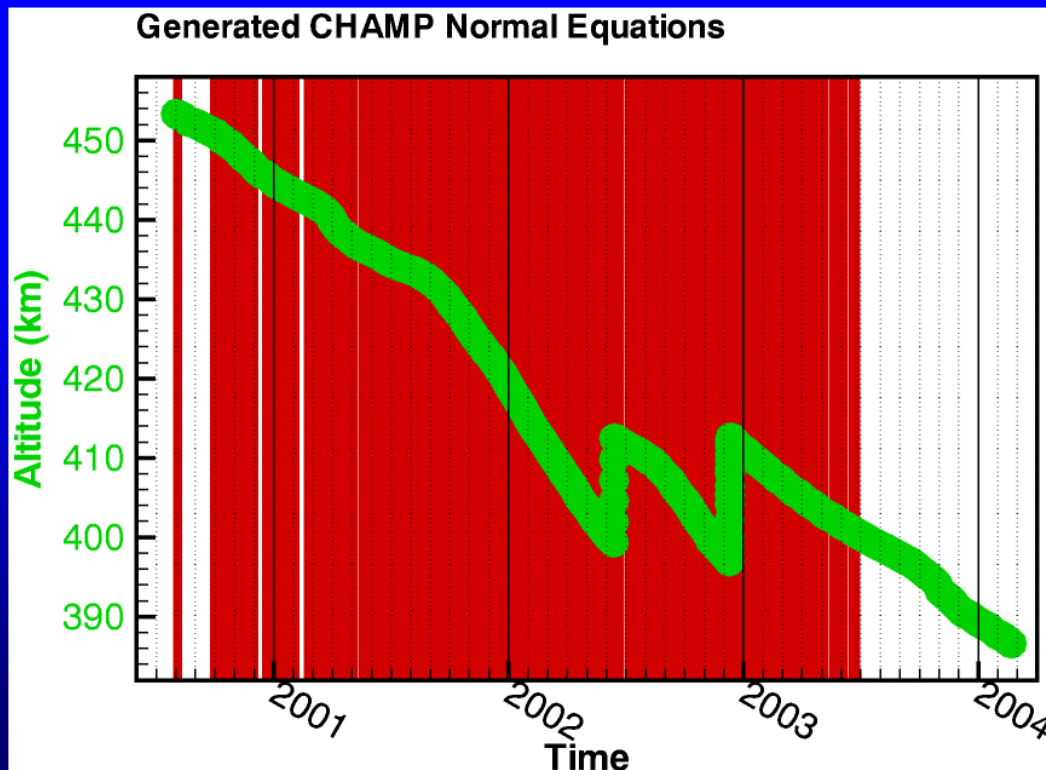
- Orbital altitude: 450 - 300 km
- Planned Mission duration: 5 years
 - Expected Mission duration: 7- 8 years

The CHAMP Mission - Gravity Recovery Part

Principle:

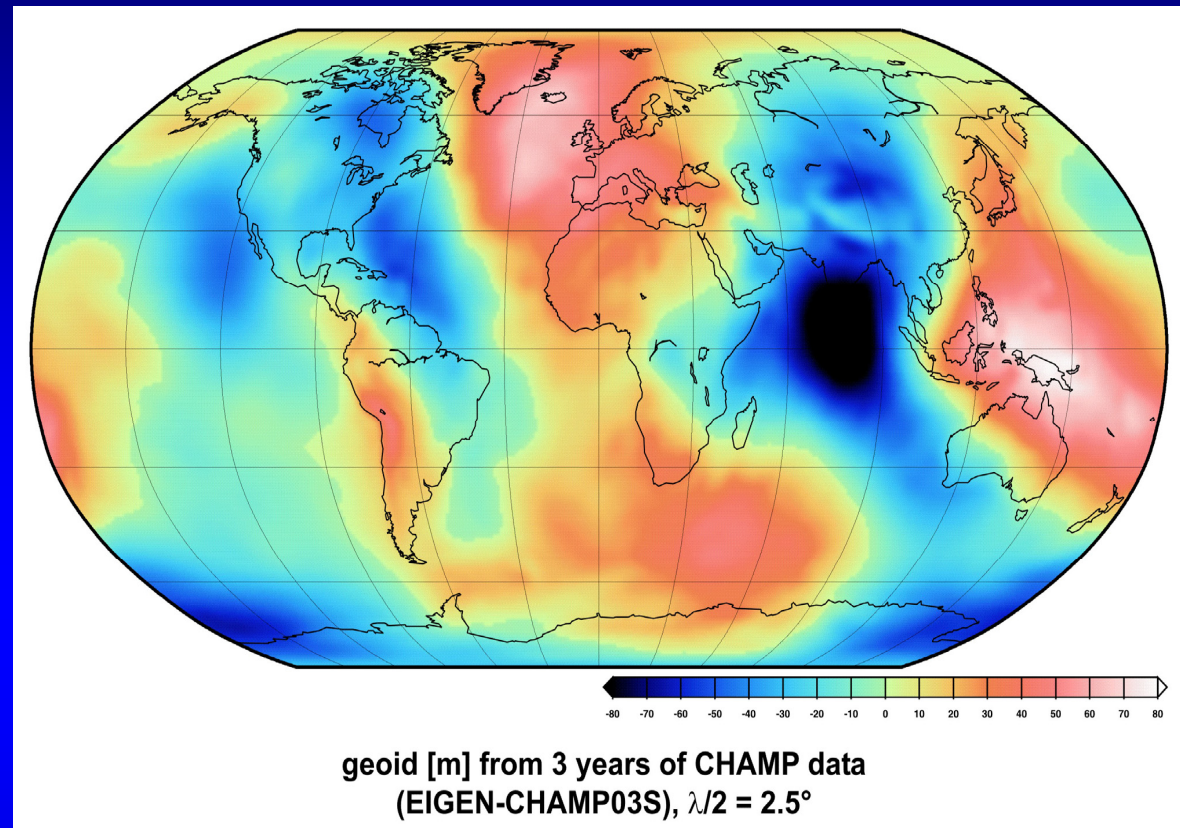
- GPS/CHAMP hi-SST-tracking
- 3D-measurement of surface forces
- Measurement of $\{x_i, \dot{x}_i\}$

Computed Normal Equations:



EIGEN – CHAMP03S GRAVITY MODEL

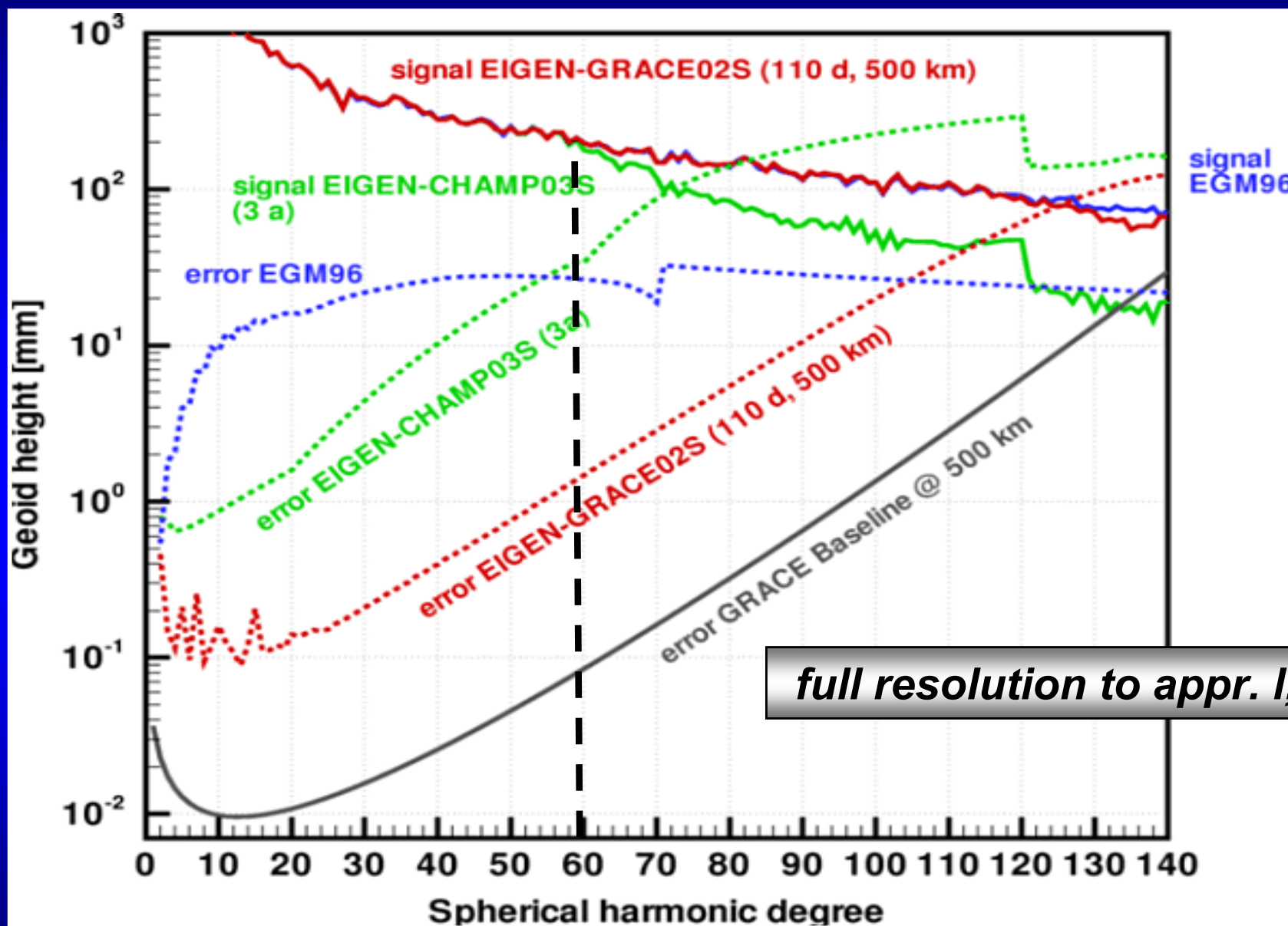
- **860 days of CHAMP data**
(07/00-07/03) (570 1.5 day arcs)
 - 12 Mio GPS – SST code & phase observations
 - Attitude from star camera
 - Non-gravitational accelerations from Star accelerometer
- **Estimated parameters**
 - Initial conditions for 1.5 day arcs
 - Clock offsets
 - GPS-SST Ambiguities
 - Accelerometer bias and scale factors
- **Estimate 120x120 field using only data from CHAMP**



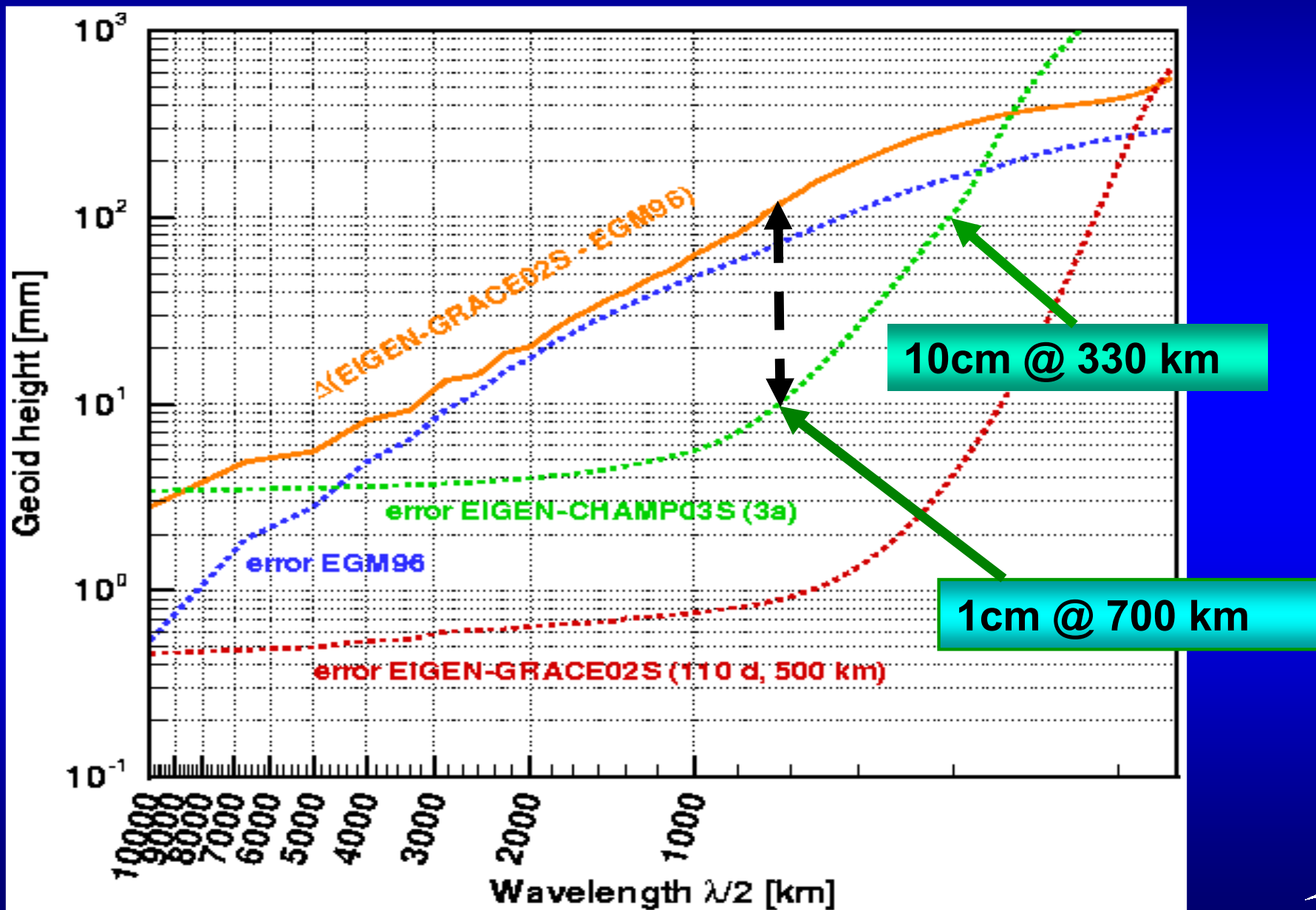
The geoid is the level (constant potential) surface that best coincides with mean sea level

The geoid height varies by ~200 m, but oceanographic applications need this to be determined to cm accuracy

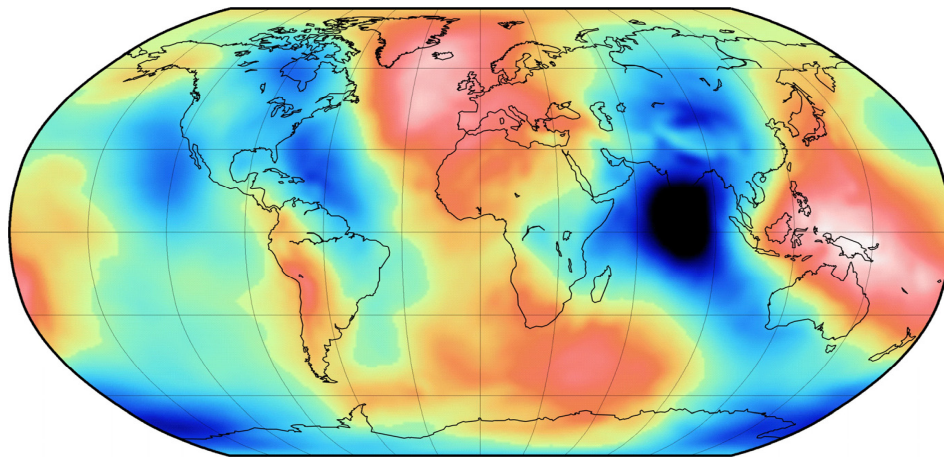
EIGEN-CHAMP03S Derived Geoid Signal & Error Spectrum



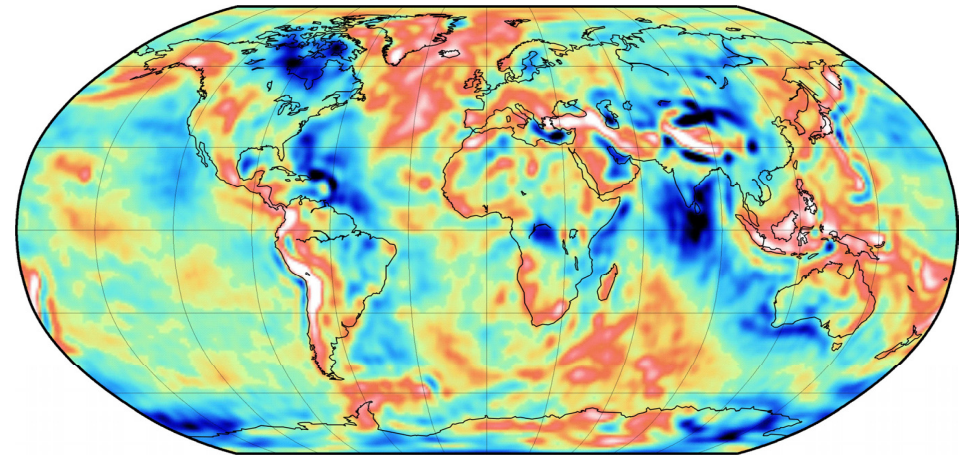
EIGEN – CHAMP03S Cumulated Geoid Error



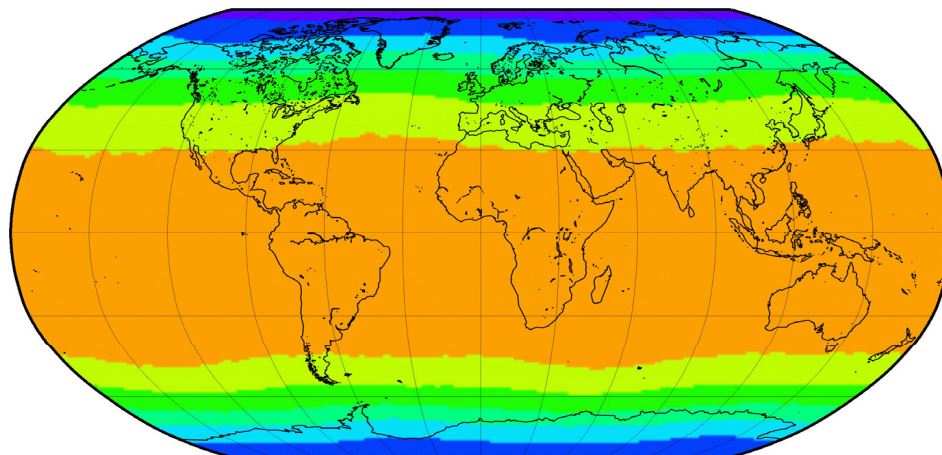
EIGEN-CHAMP03S Global Gravity Model



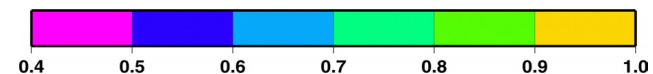
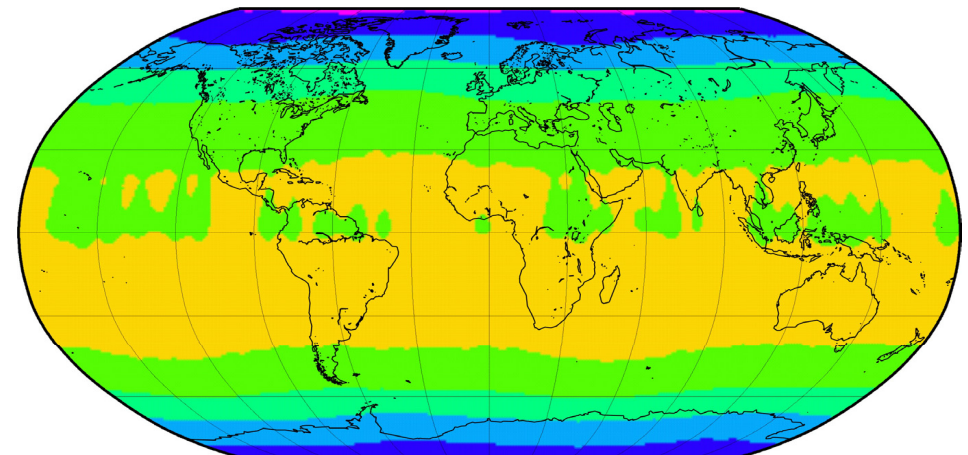
EIGEN-CHAMP03S geoid from 29 months CHAMP data, 2.5x2.5



EIGEN-CHAMP03S gravity anomalies from 29 months CHAMP data, 2.5x2.5 deg (mgal)



**EIGEN-CHAMP03S calibr. geoid standard deviations (cm),
wrms = 10.7 cm, $I_{\max} = 60$**



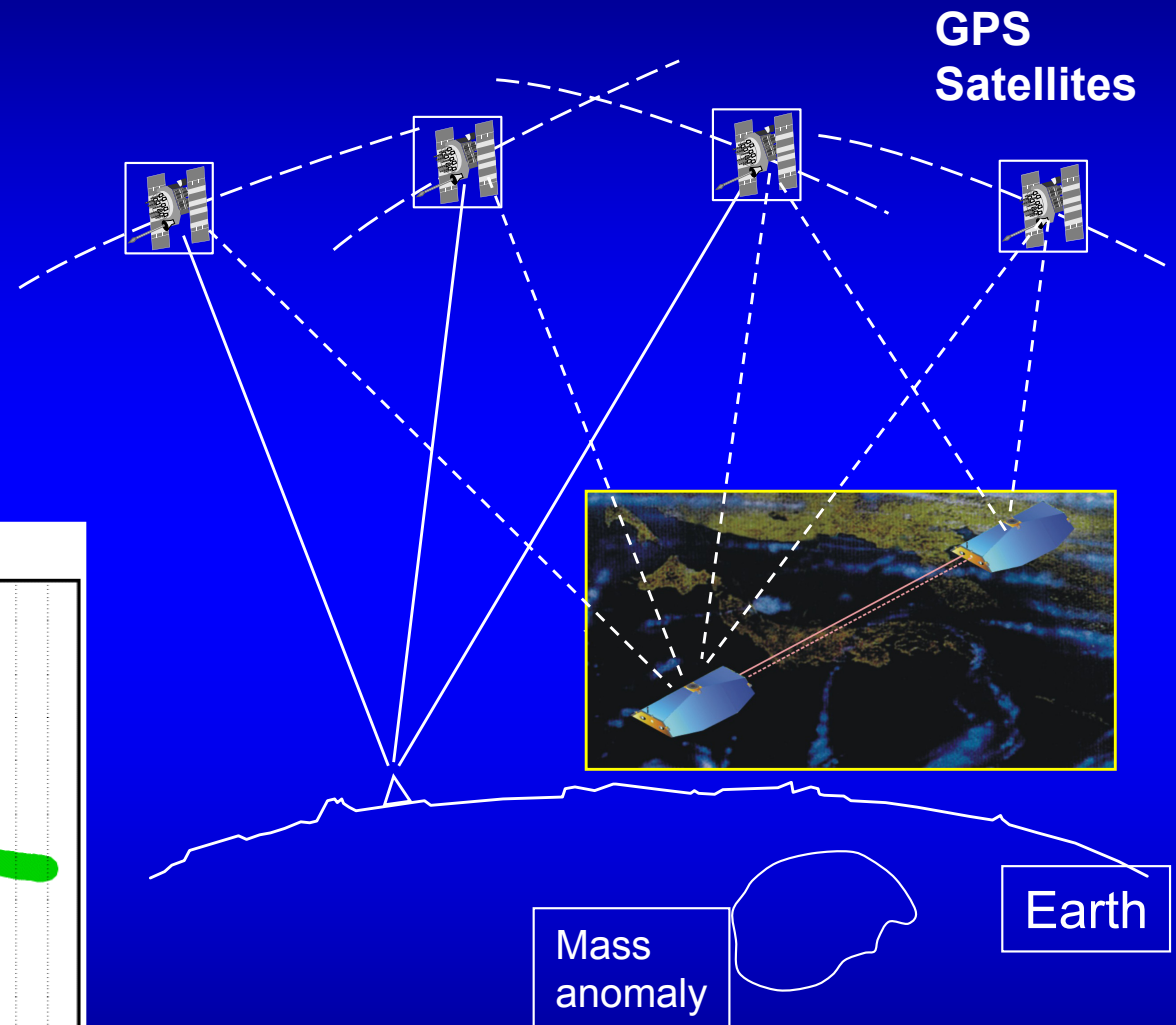
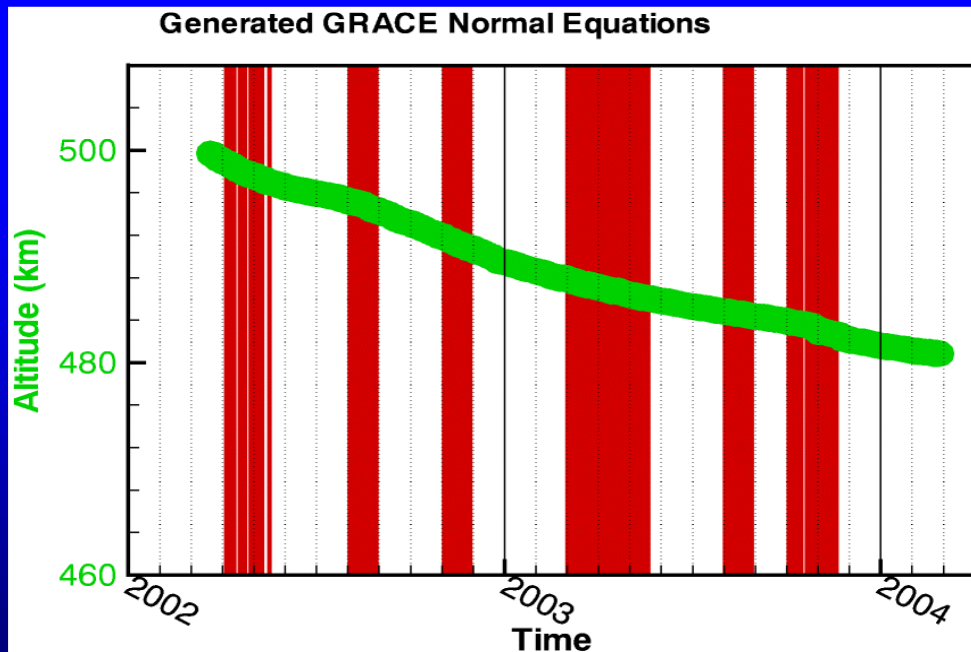
**EIGEN-CHAMP03S calibr. Δg standard deviations (mgal),
wrms = 0.86 mgal, $I_{\max} = 60$**

GRACE Mission - Gravity Recovery Part

Principle:

- GPS/GRACE hI-SST-tracking
- GRACE A/B II-SST-tracking
- 3D-measurement of surface forces
- Measurement of $\{x_i, \dot{x}_i\}$
- Measurement of $\{\rho_i, \dot{\rho}_i\}$

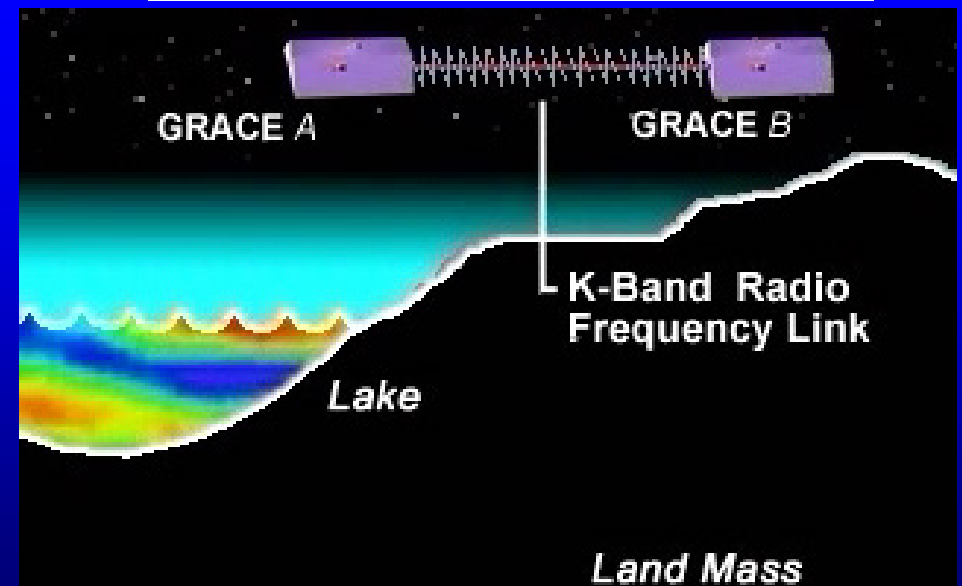
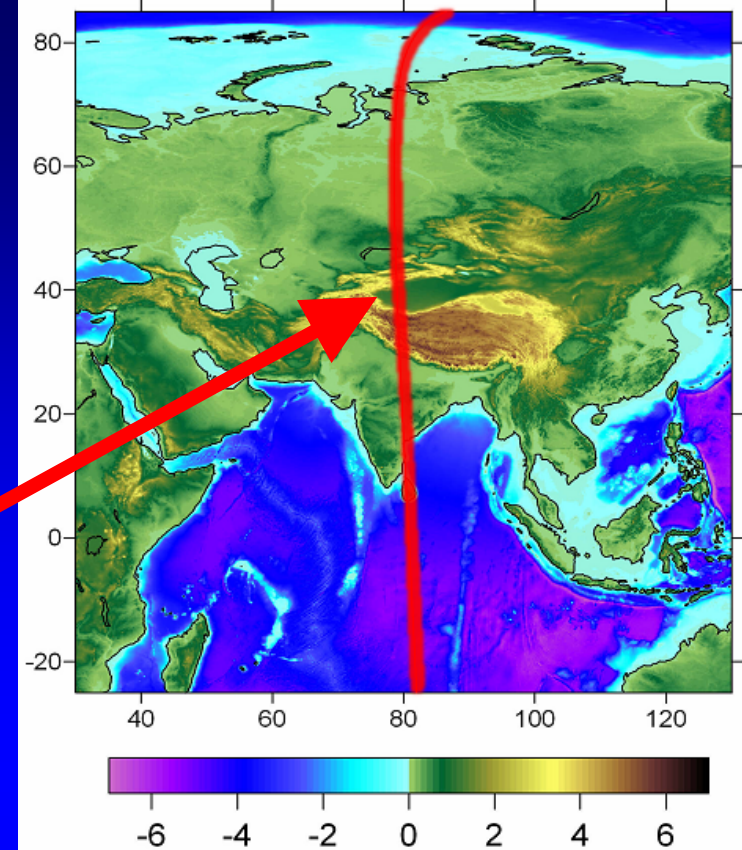
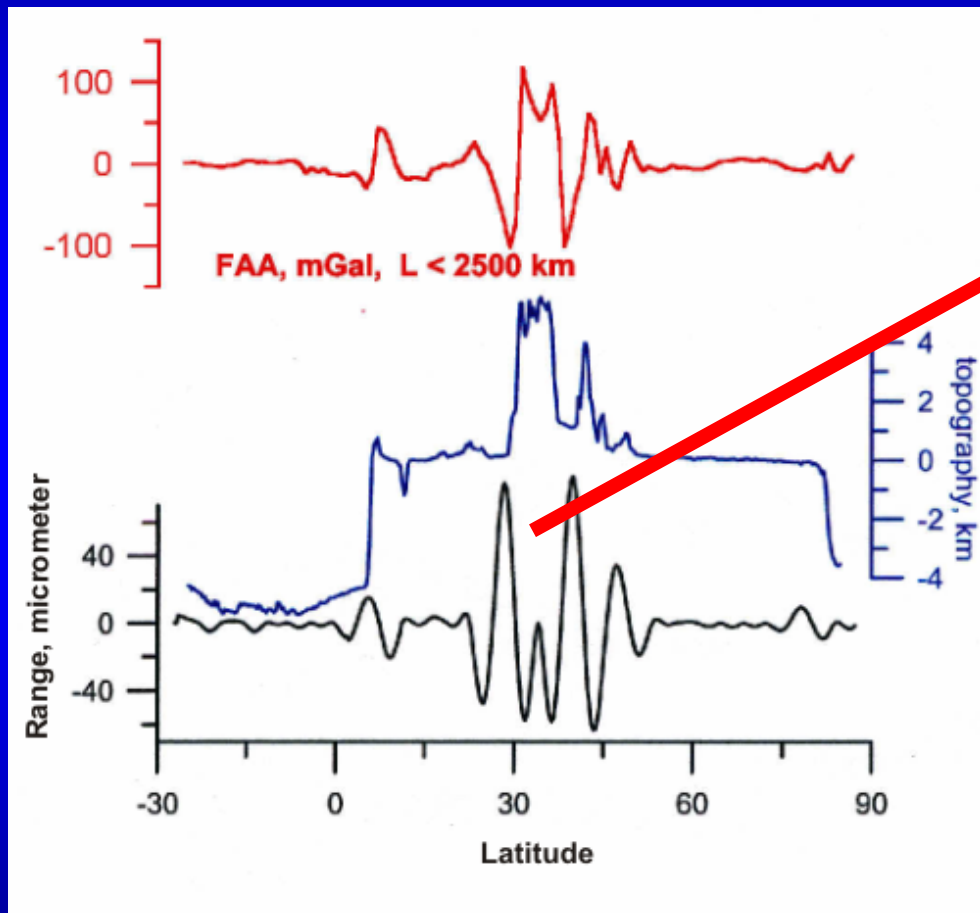
Data used so far



Source: NASA

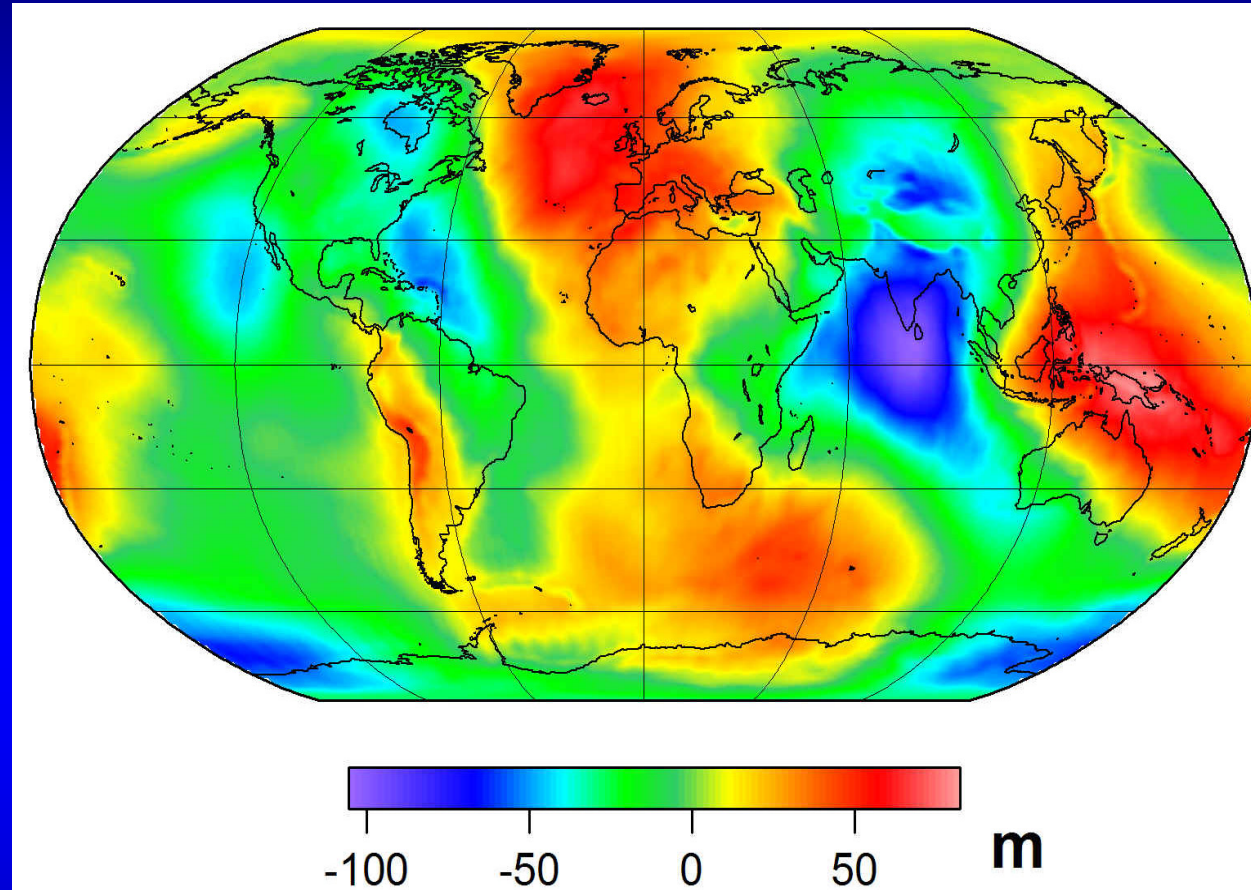
GRACE Inter-Satellite Ranging

May 3, 2003



EIGEN -GRACE GRAVITY MODEL 02

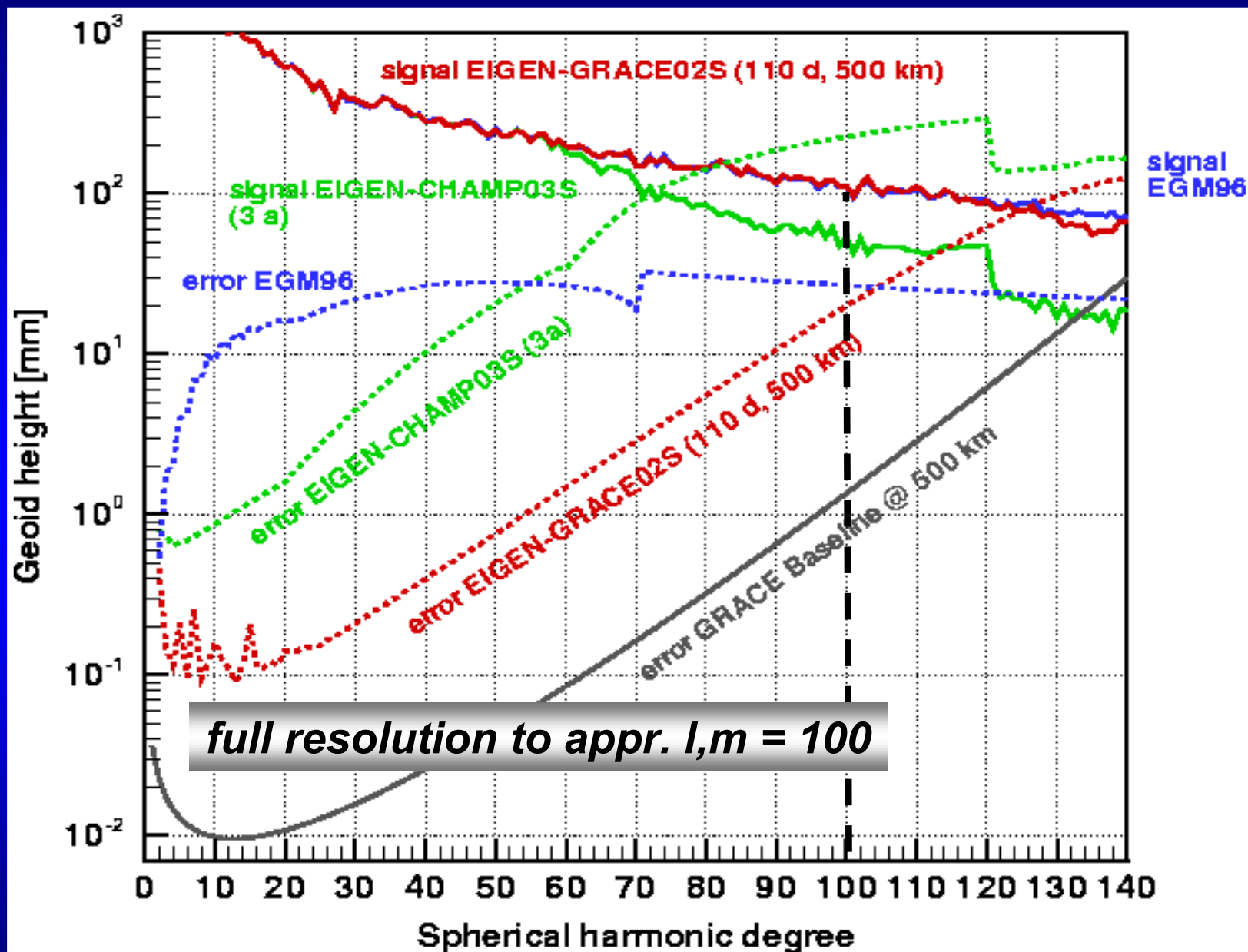
- **109 days of GRACE data**
(08/02-04/03)
 - KBR range-rate and GPS phase data
 - Attitude from star camera
 - Non-gravitational accelerations from SuperStar accelerometer
- **Estimated parameters**
 - Initial conditions for daily arcs
 - Accelerometer bias and scale factors
 - KBR biases
- **Estimate 150x150 using only data from GRACE**



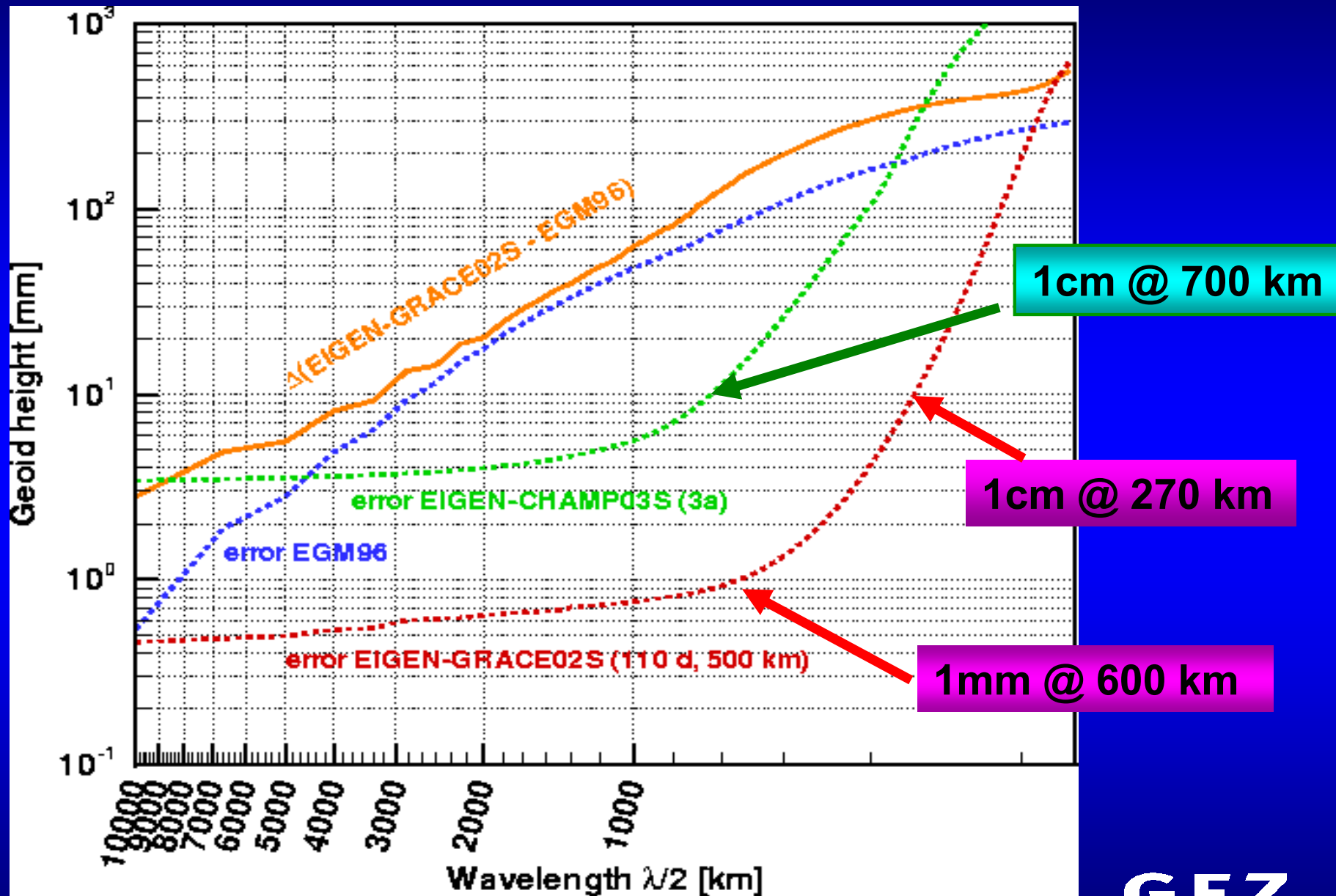
The geoid is the level (constant potential) surface that best coincides with mean sea level

The geoid height varies by ~200 m, but oceanographic applications need this to be determined to cm accuracy

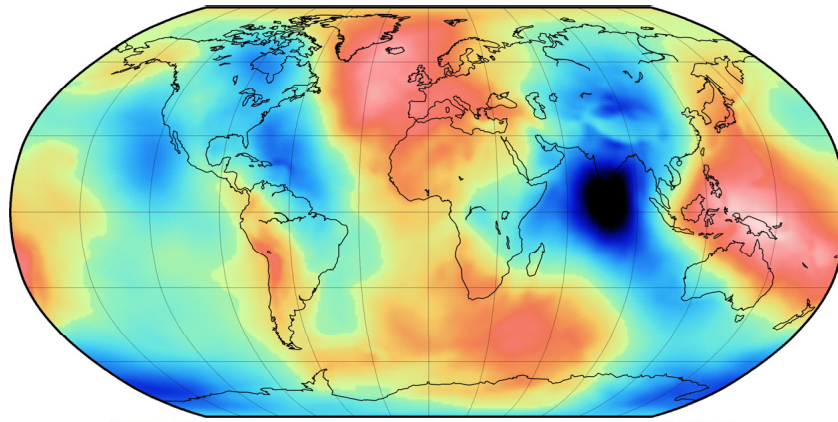
EIGEN-GRACE02S Derived Geoid Signal & Error Spectrum



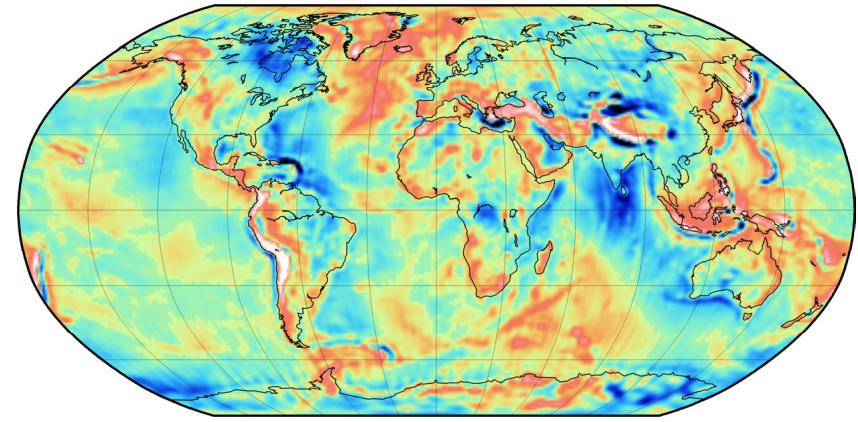
EIGEN – GRACE02S Cumulated Geoid Error



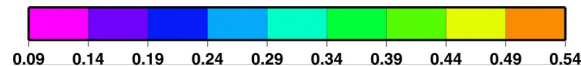
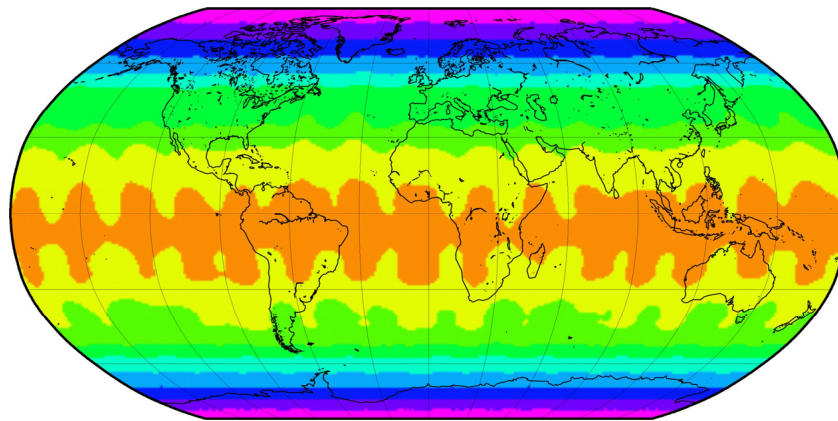
EIGEN-GRACE02S Global Gravity Model



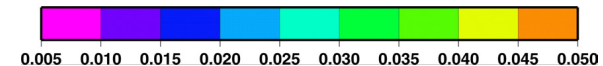
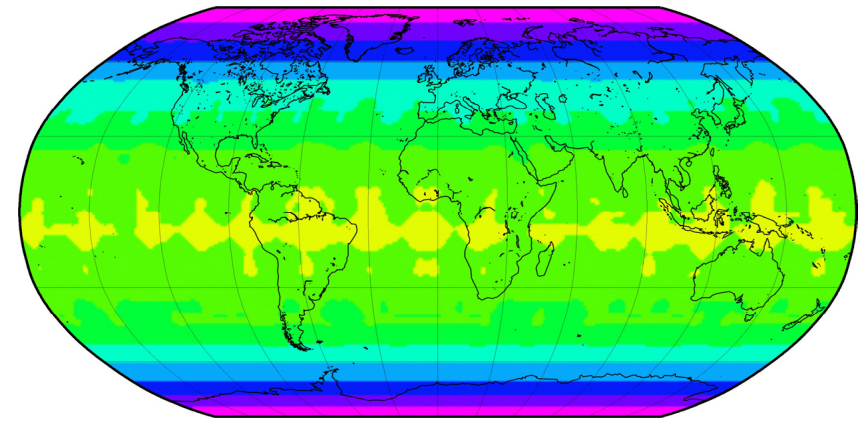
EIGEN-GRACE02S geoid from 39 days
GRACE data, 1.5x1.5 deg (m)



EIGEN-GRACE02S gravity anomalies from 39 days
GRACE data, 1.5x1.5 deg (mgal)



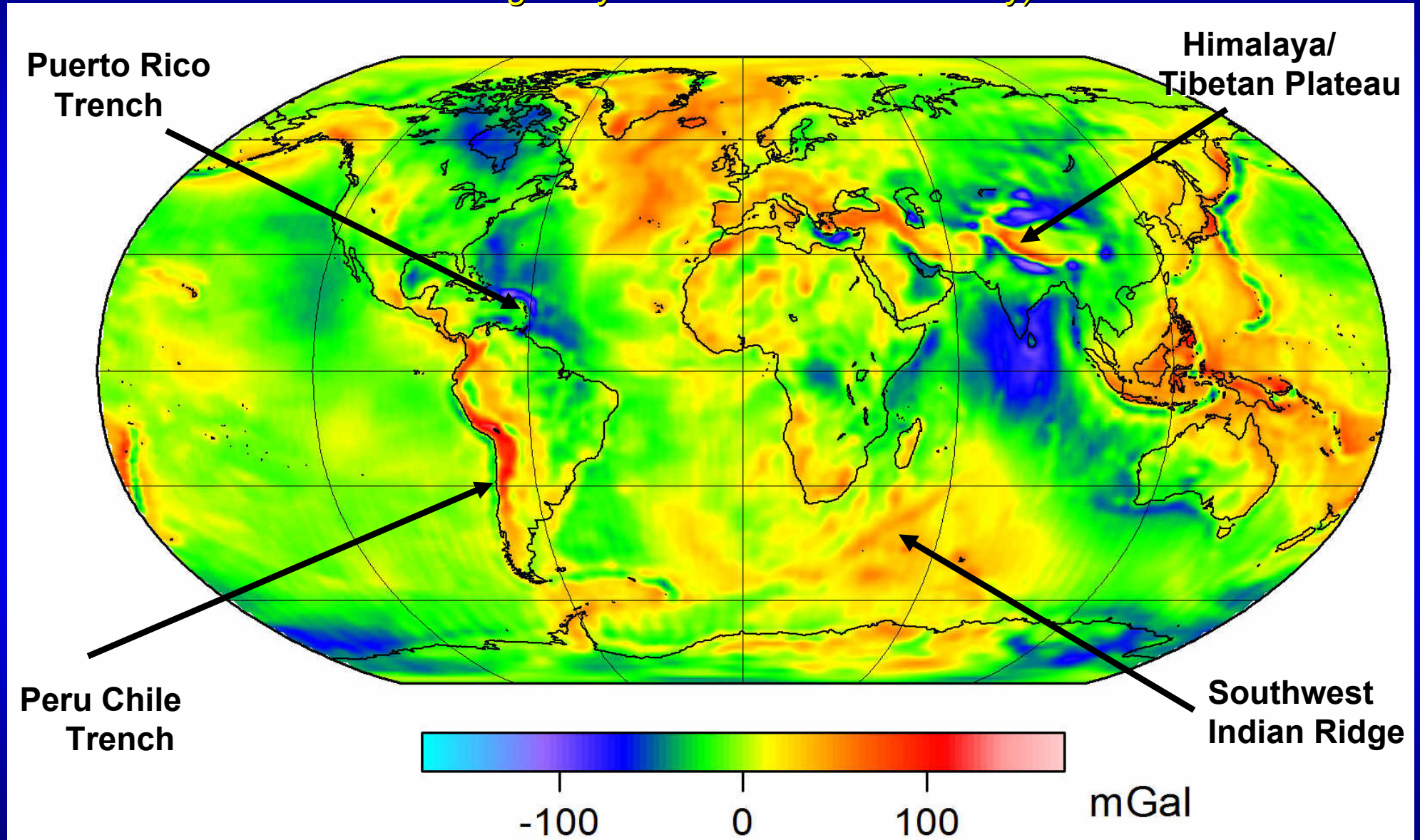
EIGEN-GRACE02S calibr. geoid standard deviations (cm), EIGEN-GRACE02S calibr. Δg standard deviations (mgal),
wrms = 0.42 cm, $I_{\max} = 60$



wrms = 0.034 mgal, $I_{\max} = 60$

Gravity Field as Seen from Space by GRACE

(only 109 days from first science data set used,
no surface gravity and no satellite altimetry)



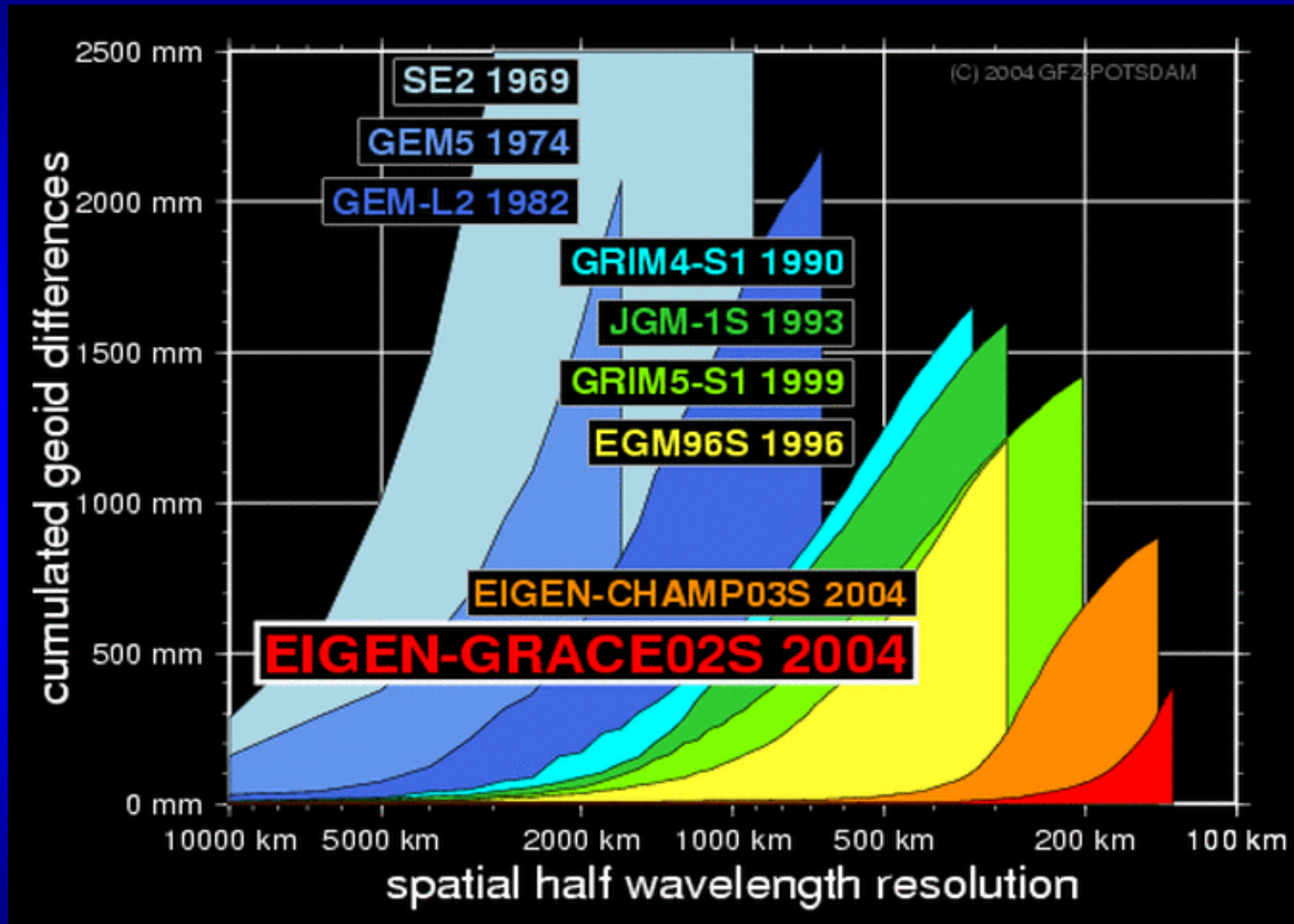
Orbital Fit Results for Various Geodetic Satellites

SLR resp. GPS Phase/SLR RMS (CHAMP, GRACE)

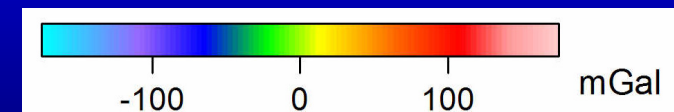
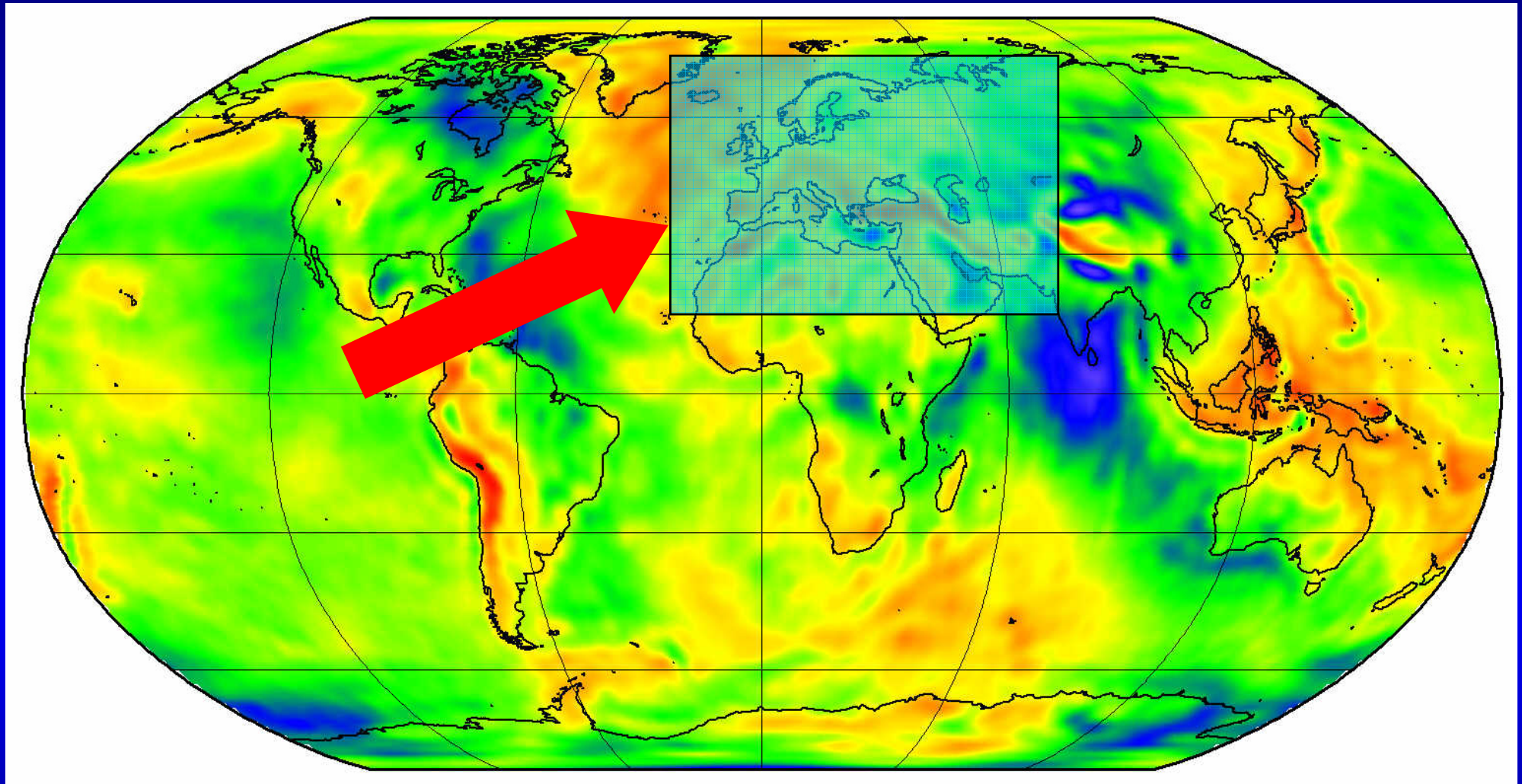
Gravity Model	Starlette [cm]	Stella [cm]	Lageos-1/2 [cm]	GFZ-1 [cm]	Ajis [cm]	ERS-2 [cm]	ENVI [cm]	CHAMP [cm]	GRACE [cm]
<i>MultiSatComb</i>									
GRIM5-C1	2.7	3.1	1.11/1.06	14.7	3.3	5.5	4.5	10.4/62	36.4/217
EGM96	3.2	6.7	1.15/1.15	24.7	4.0	9.2	7.1	10.2/81	19.1/70
TEG4*)	3.7	3.4	1.11/1.09	20.5	3.5	5.7	5.3	1.4/14	5.1/28
<i>CHAMP-only</i>									
EIGEN-3	3.4	6.8	1.15/1.08	13.6	3.4	7.3	15.2	0.6/5.7	2.2/10.1
<i>GRACE-only</i>									
GGM01S-CSR	2.5	3.6	1.13/1.05	14.5	3.3	5.9	6.3	0.6/6.7	1.4/6.7
GRACE01S	2.7	3.4	1.11/1.05	13.5	3.5	6.0	4.8	0.6/6.0	1.3/7.5
GRACE01Sup	2.5	3.5	1.11/1.05	14.5	3.1	5.7	5.2	0.6/6.9	1.2/6.0
GRACE02S	2.7	3.2	1.13/1.07	14.3	3.2	5.5	4.5	0.5/5.8	1.1/5.6

All arcs are not included in gravity modelling

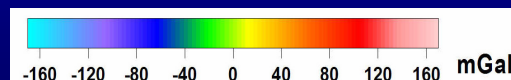
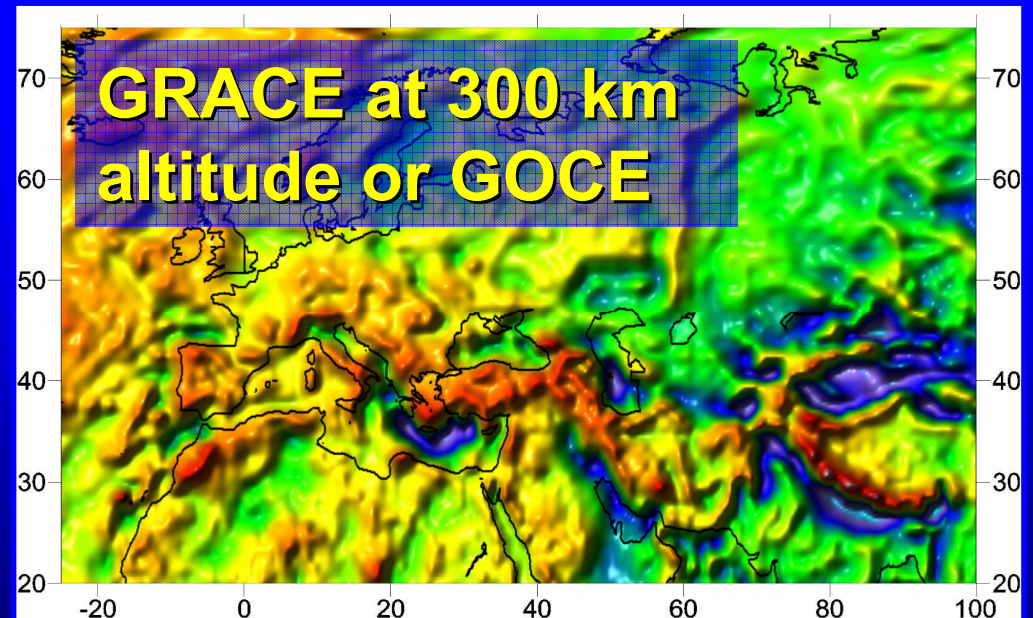
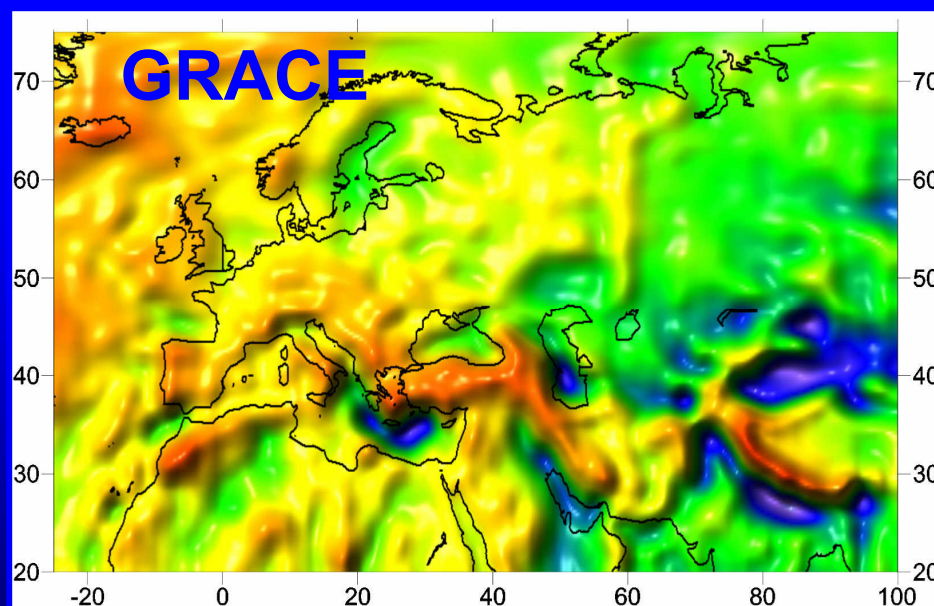
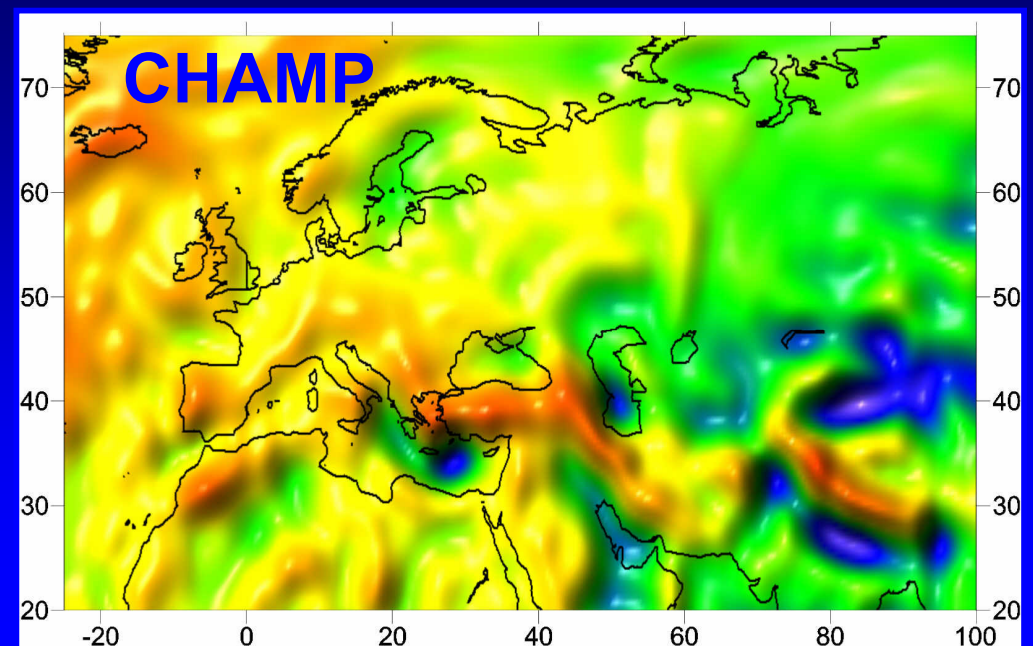
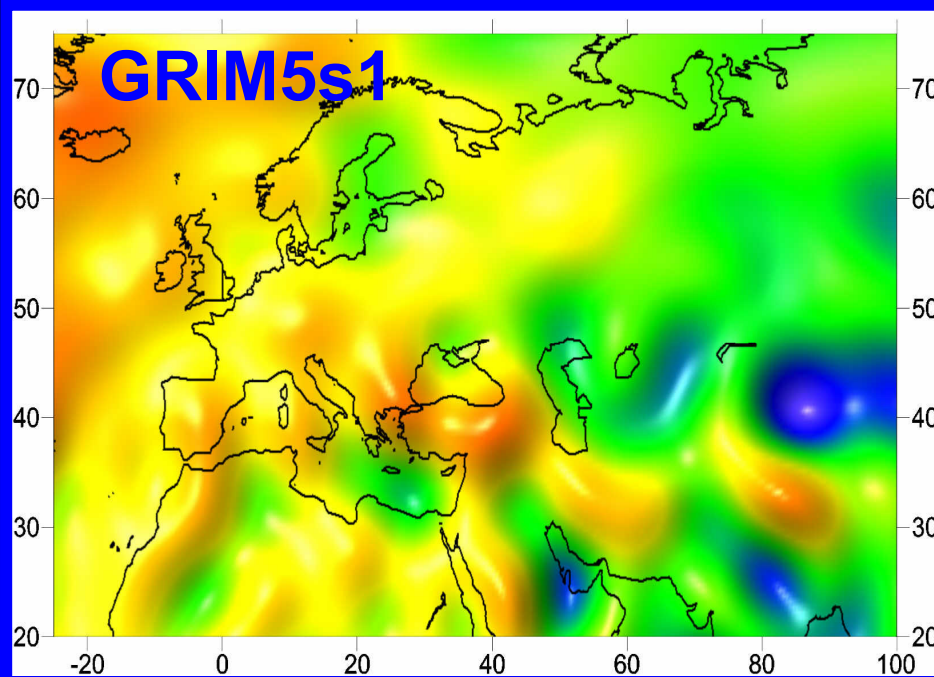
Geoid Differences to GRACE Combined Solution

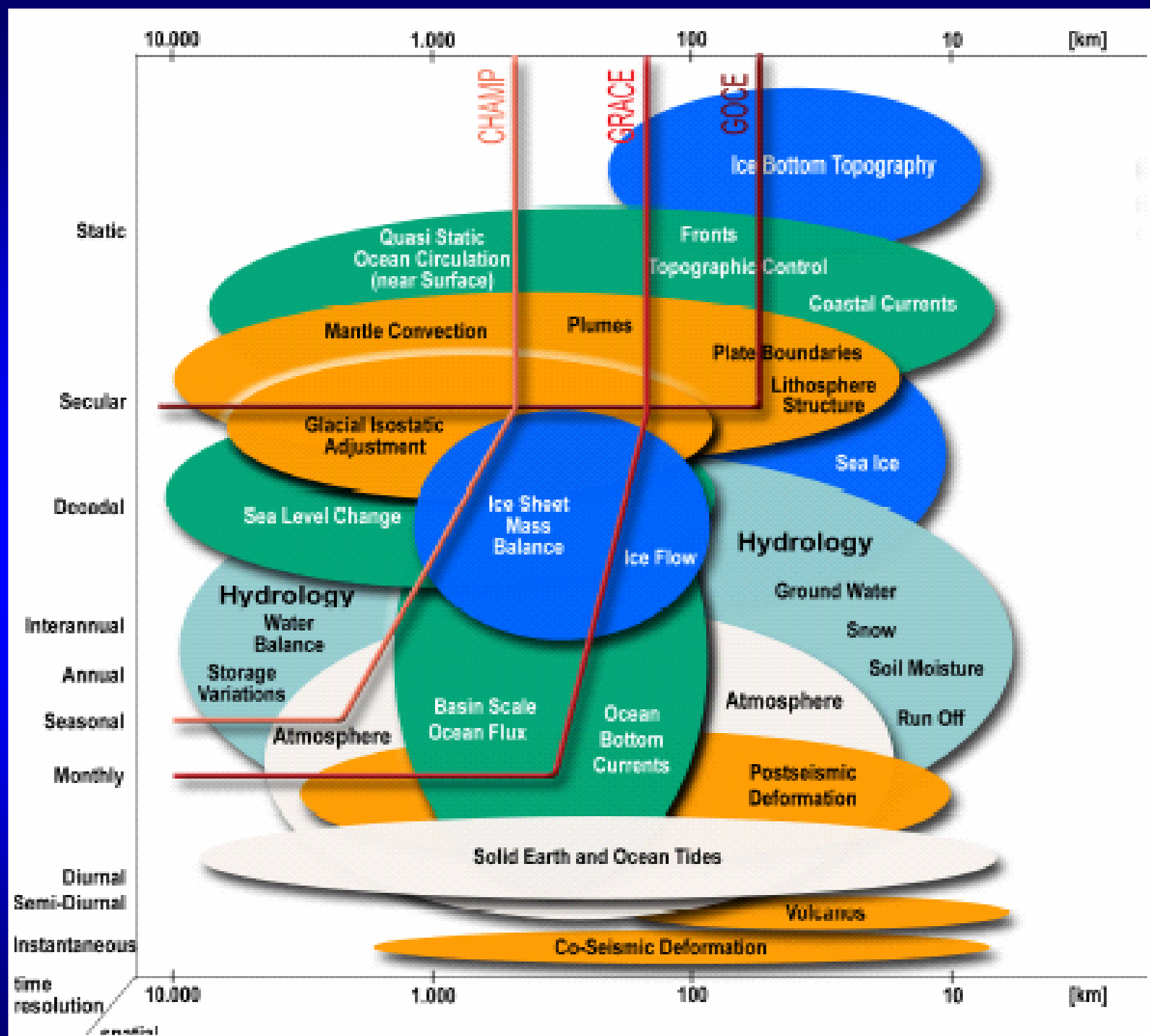


Detailed View of Gravity over Europe



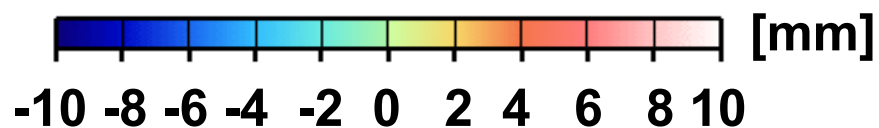
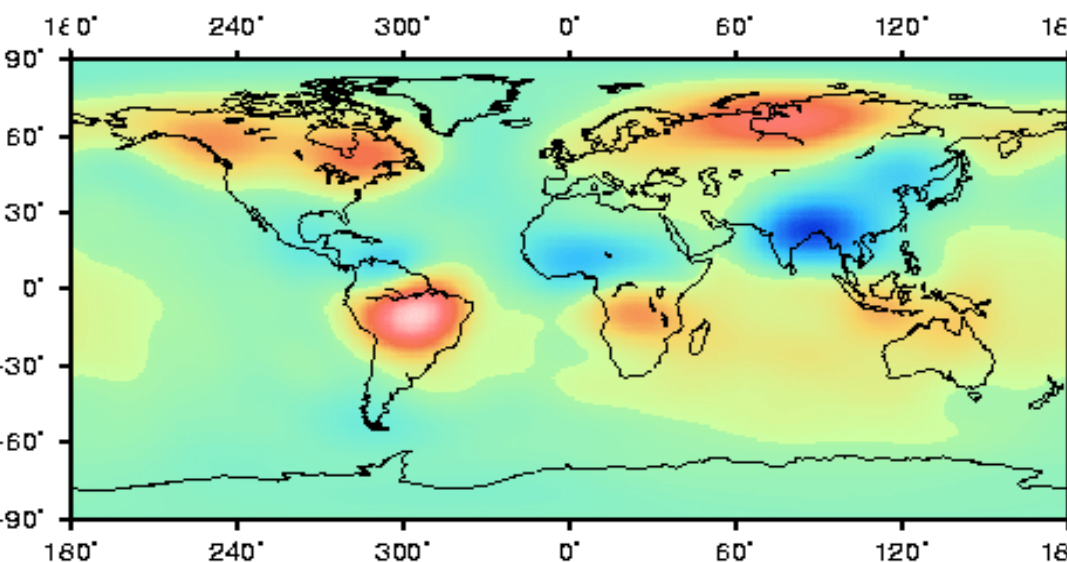
Gravity over Europe as Seen by Different Missions



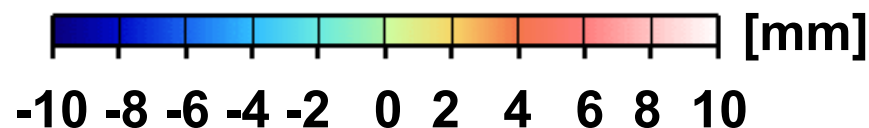
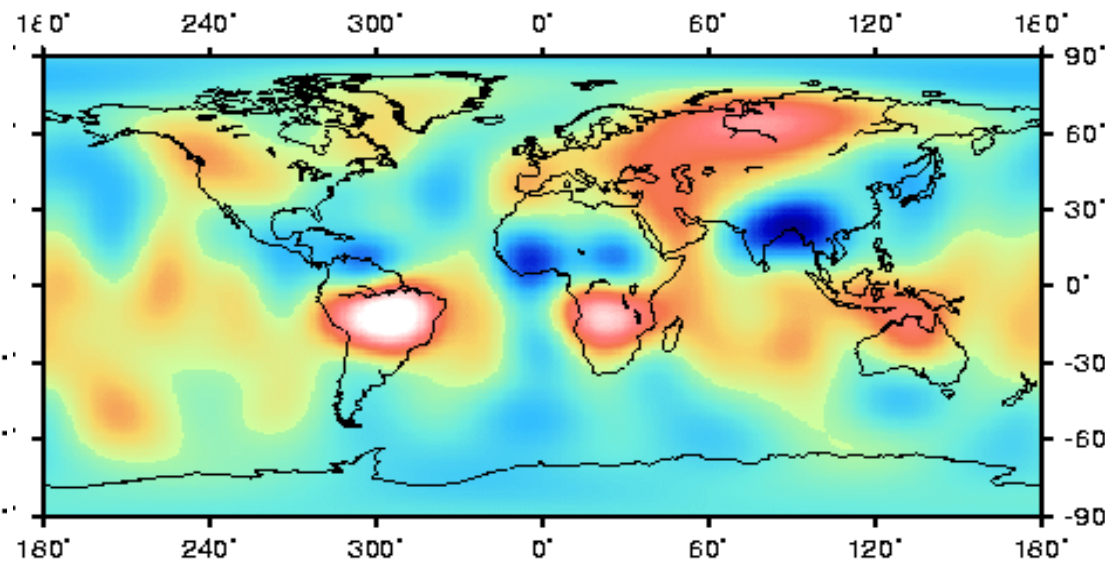


MAI03 - AUG03 ($N_{max} = 13$)

Hydrologie - WGHM

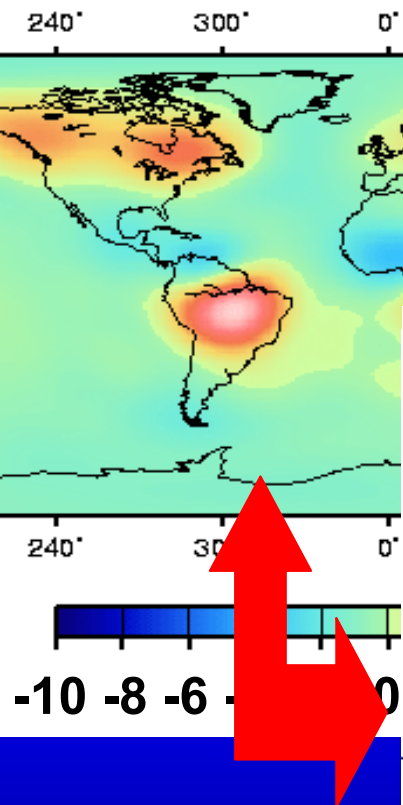


GRACE

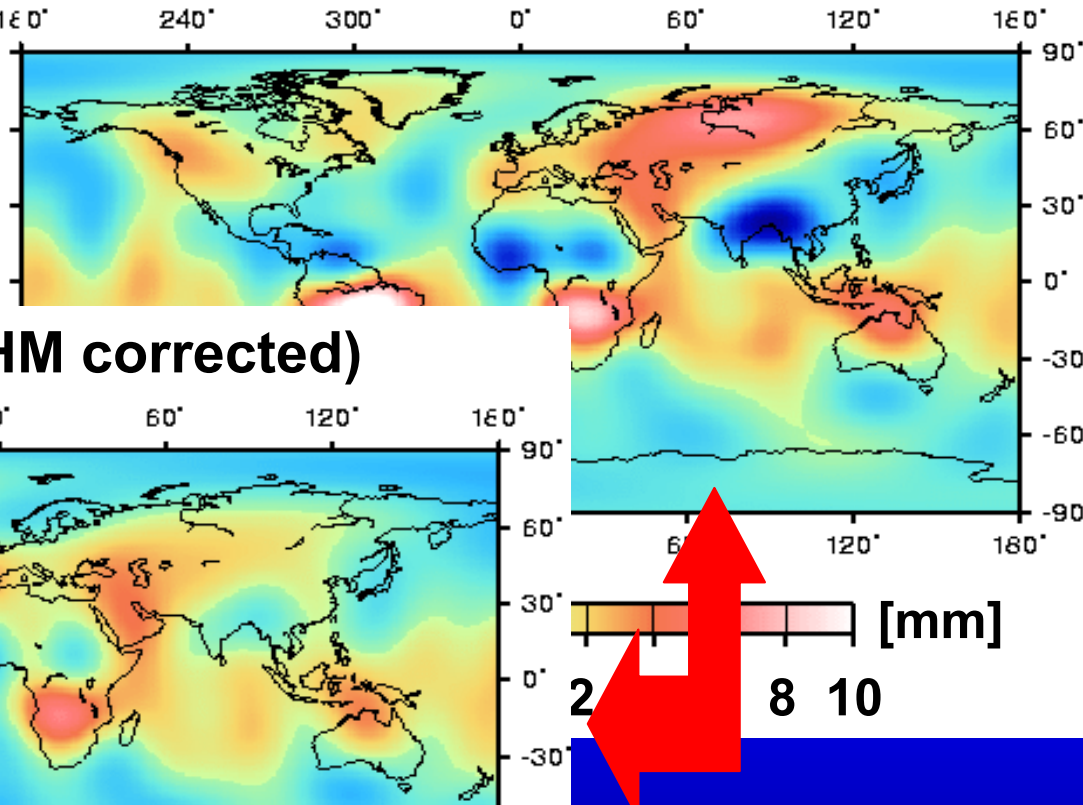


MAI03 - AUG03 ($N_{max} = 13$)

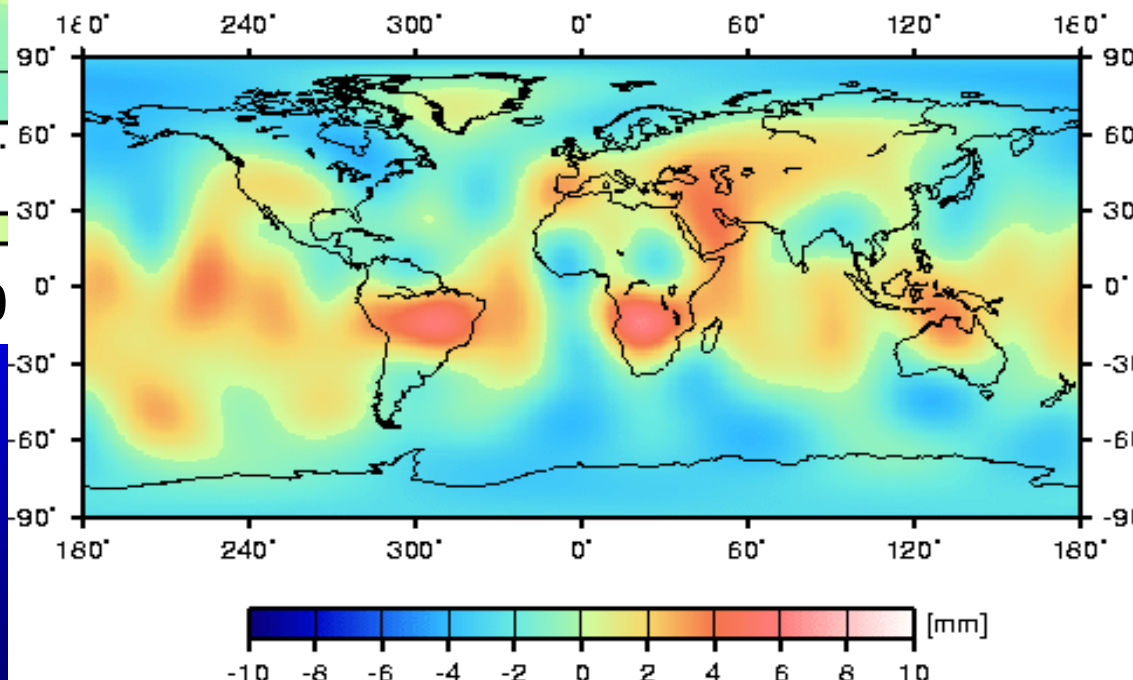
Hydrologie - WGHM



GRACE



GRACE (WGHM corrected)



Correlation GRACE - Hydrology

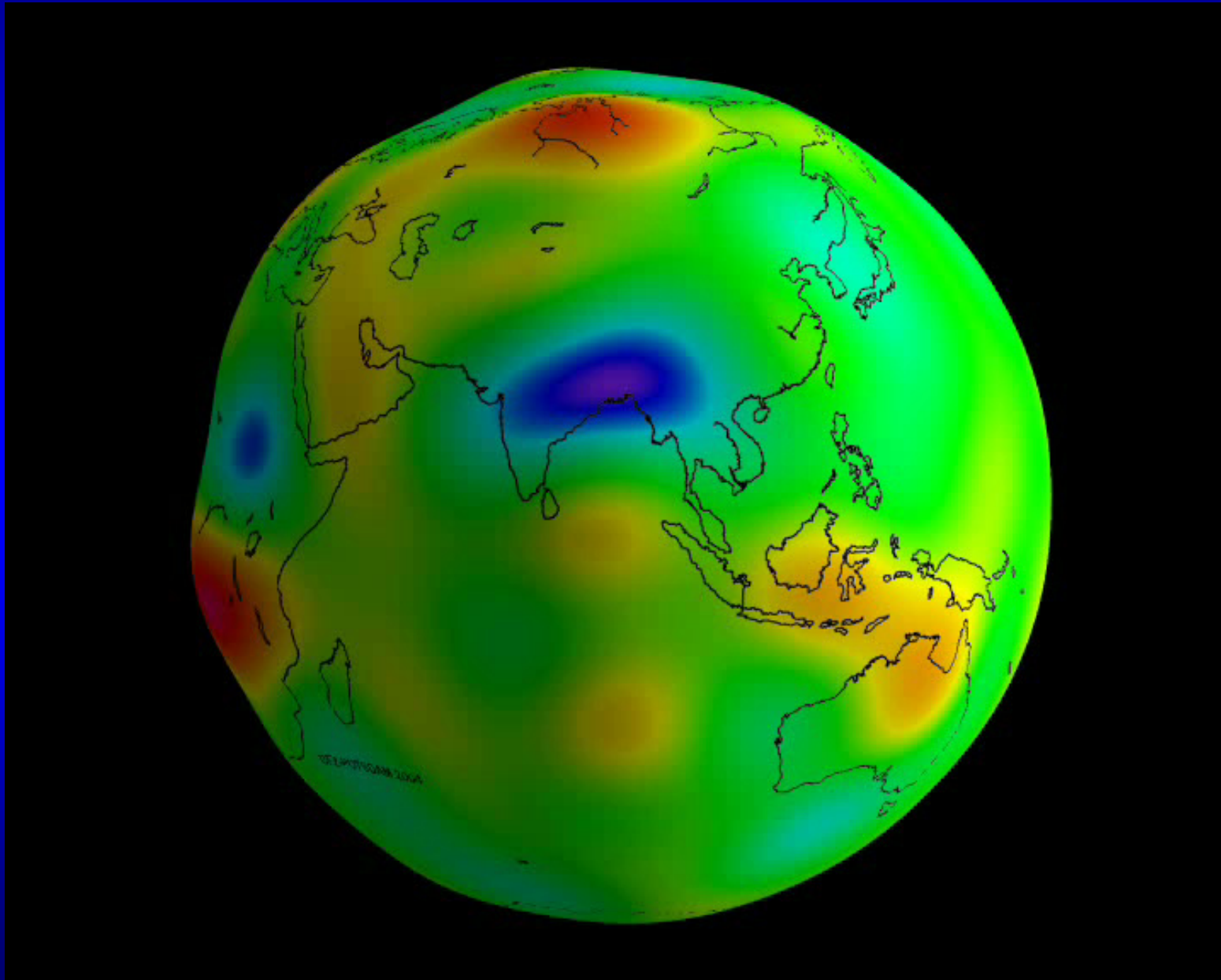
Difference	Δt	WGHM	Milly	Yun-Fan
APR02 - MAY02	1 M	0.47	0.52	0.60
MAY02 - AUG02	3 M	0.76	0.77	0.63
AUG02 - NOV02	3 M	0.47	0.25	0.56
NOV02 - MAR03	4 M	0.84	0.90	0.80
MAR03 - APR03	1 M	0.28	-0.14	0.19
APR03 - MAY03	1 M	0.66	0.51	0.38
MAY03 - AUG03	3 M	0.82	-	0.89

Correlation Geoid Height Variation from GRACE and Hydrology Models on 1x1 grid over land

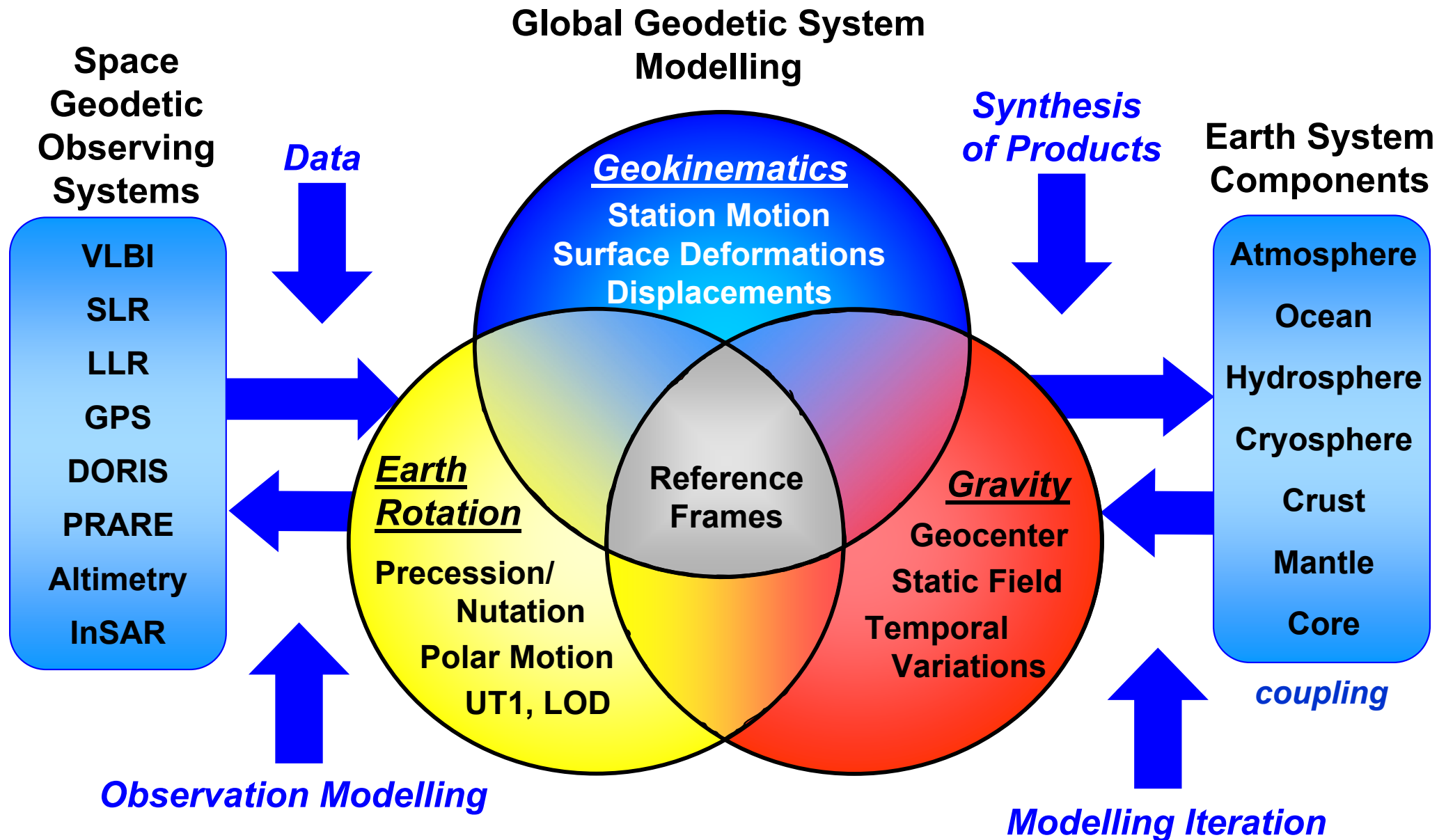
Variation in Continental Water Storage

from GRACE monthly gravity solutions (nmax=13)

(here April 03 – August 02)



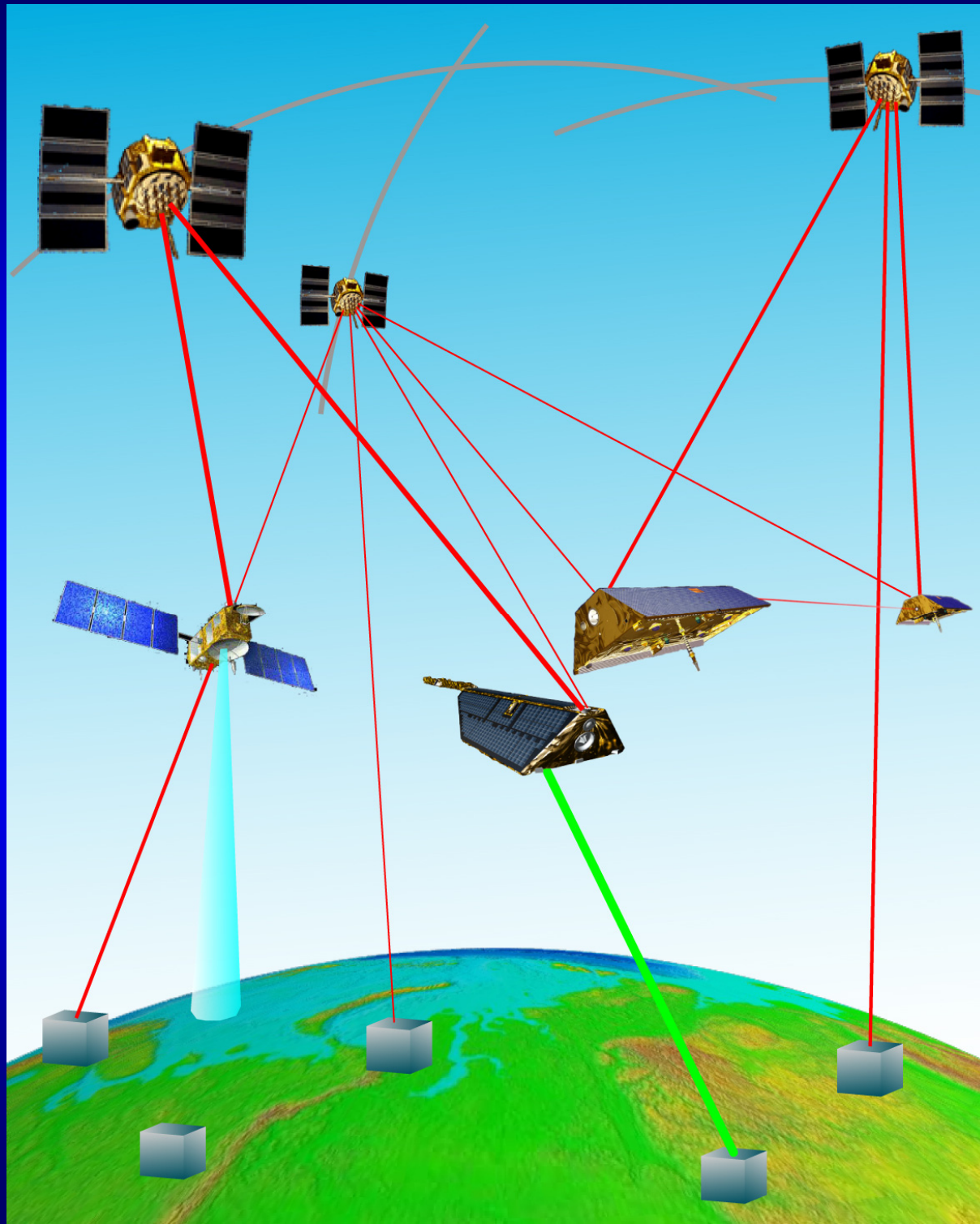
Observing and Modelling the Earth System



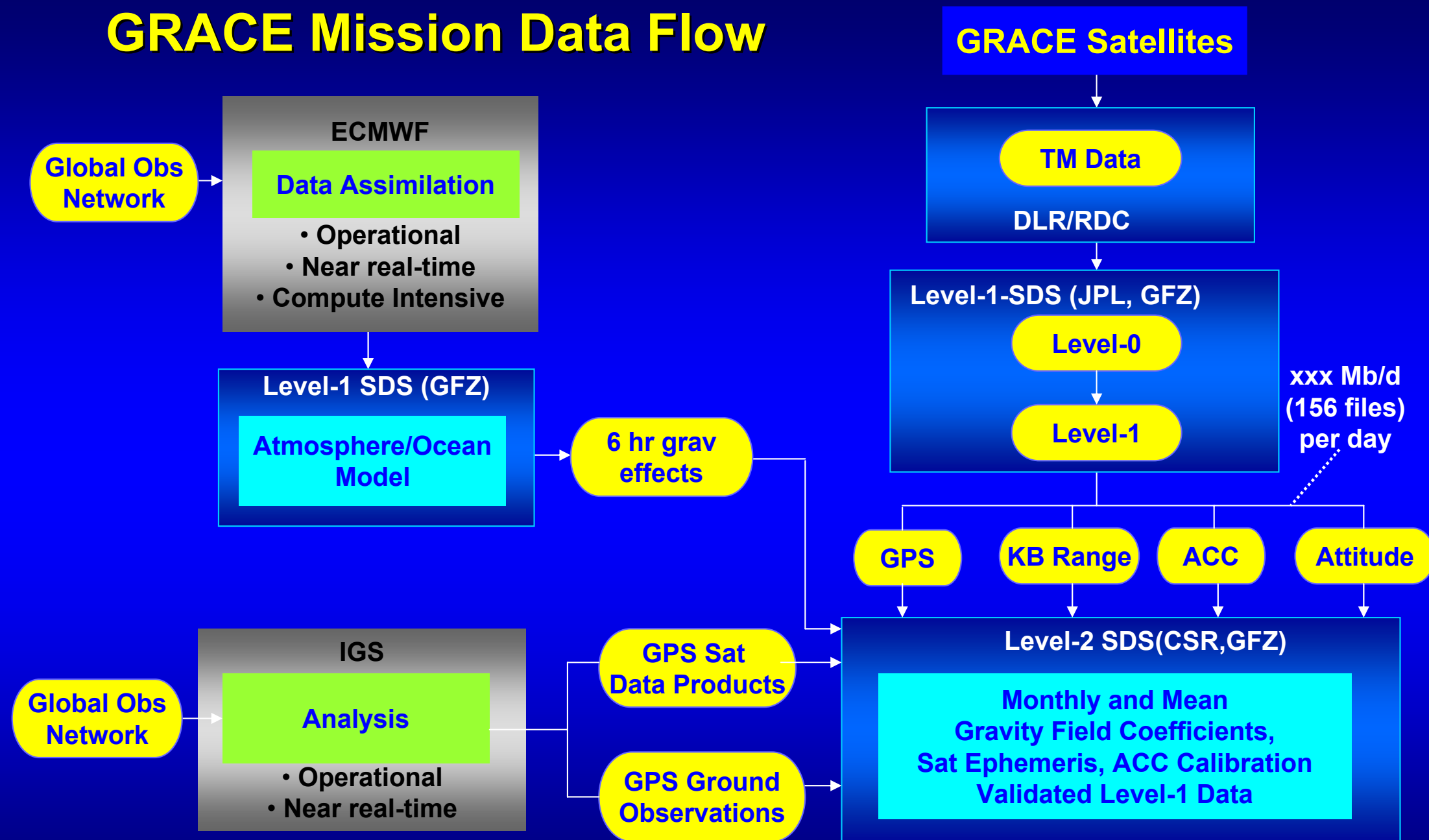
Layer 3

Layer 2

Layer 1



GRACE Mission Data Flow



GRACE Mission Data Flow

GRACE Satellites

Global Obs
Network

ECMWF

Data Assimilation

- Operational
- Near real-time
- Compute Intensive

Level-1 SDS (GFZ)

Atmosphere/Ocean
Model

Thank You IGS Community
for the Great Support of
both the CHAMP and the
GRACE Mission

Global Obs
Network

IGS

Analysis

- Operational
- Near real-time

GPS Sat
Data Products

GPS Ground
Observations

Level-2 SDS(CSR,GFZ)

Monthly and Mean
Gravity Field Coefficients,
Sat Ephemeris, ACC Calibration
Validated Level-1 Data



